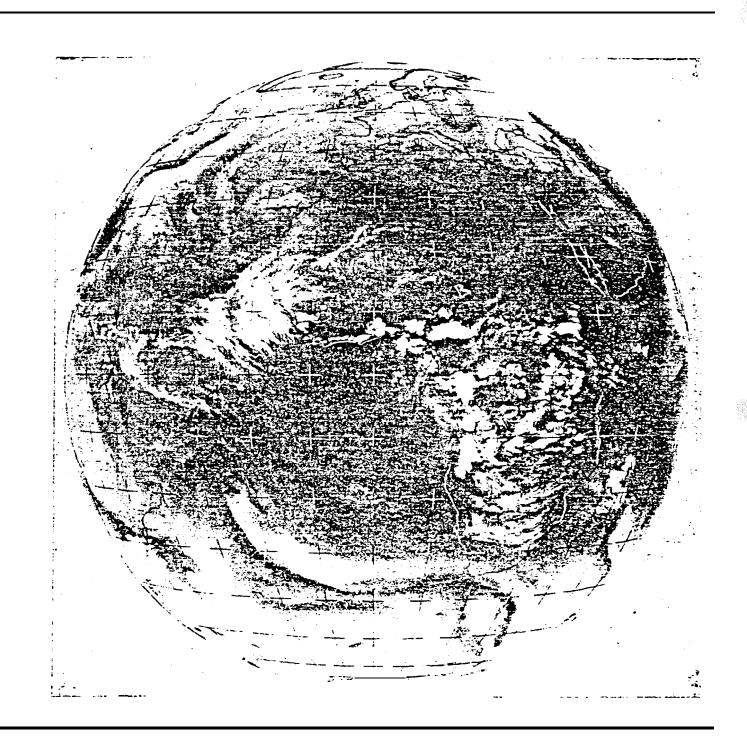


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M ETEOSAT HIGH RESOLUTION IMAGE DISSEMINATION



METEOSAT 'SYSTEM GUIDE - VOLUME 9

METEOSAT HIGH RESOLUTION IMAGE DISSEMINATION

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ISSUE 4 .
AUGUST, 1984

Ref : M.S.9.0.1.1 Volume 9 Section 0 Issue 4 Valid from 01.09.1984

TABLE FOR NOTING AMENDMENTS RECEIVED

Amendment No.	Dated	Inserted in the publication by date
1	15/12/86	
2		
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17		
18		
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Volume 9 Section 0

Issue 4 Valid from 01.09.1984

METEOSAT HIGH RESOLUTION IMAGE-DISSEMINATION

CONTENTS

Introduction

1.	Dissemination of Images
2.	Primary Data User Stations
3.	Image Sources
3.1	METEOSAT
3.2	GOES-E
	•
4.	High Resolution Formats
4.1	HR-Formats originated in ESOC, Darmstadt
4.1.1	A-Format
4.1.2	B-Format
4.2	HR-Format originated in CMS Lannion
4.2.1	X-Format
5.	Digital Transmission Format `
5.1	Definition of Terms
5.2	Format Construction
5.3	Elements of Subframes
5.3.1	Synchronisation Word
5.3.2	ID-Word
5.3.3	Label
5.3.4	Identification
5.3.5	Interpretation Data

5.3.6	Fillers			
5.3.7	Grid Information			
5.4	Construction of Subframes			
5.4.1	Heading Subframes			
5.4.1.1	Heading Subframes of A-Formats			
5.4.1.2	Heading Subframes of B- and X-Formats			
5.4.2	Data Subframes			
5.4.2.1	A-Format Data Subframes			
5.4.2.2	B- and X-Format Data Subframes			
5.4.3	Annotation Subframes			
5.4.4	Conclusion Subframes			
5.5	Sets of Data Subframes			
5.6	Transmission Formats			
5.7	File Layout of Interpretation Data			
5.7.1	Calibration Data			
5.7.2	Spacecraft Operations Data .			
5.7.3	Imagery Data			
5.7.4	Administrative Messages			
5.8	Content of Interpretation Data			
5.8.1	Calibration Data			
5.8.1.1	Normalised Black Body Count			
5.8.1.2	Standard Deviaton for BBC			
5.8.1.3	Timestamp for BBC			
5.8.1.4	IR-Calibration Value			
5.8.1.5	Timestamp for IR Calibration Value			
5.8.1.6	Water Vapour Calibration Value			
5.8.1.7	Timestamp for WV-Calibration Value			
5.8.1.8	Gains			
5.8.2	Spacecraft Operations Data			
5.8.2.1	Right Ascension Attitude South			
5.8.2.2	Declination Attitude South			
5.8.2.3	Right Ascension Attitude North			

Volume 9 Section 0 Issue 4 Valid from 01.09.1984

F 0 0 4	Dealination Attitude Newth			
5.8.2.4	Declination Attitude North			
5.8.2.5	X, Y, Z of Refined Attitude			
5.8.2.6	Right Ascension and Declination			
5.8.2.7	Number of Slots for Refined Attitude			
5.8.2.8	Spin Duration			
5.8.2.9	Flags for Special Operation Modes			
5.8.3	Imagery Data			
5.8.3.1	Image and Processing Status Information			
5.8.3.2	Southern Line etc.			
5.8.3.3	Distance Satellite-Earth Centre			
5.8.3.4	X, Y, Z Components			
5.8.3.5	Max. Deformation Differences			
5.8.3.6	Image Conditions			
5.8.3.7	Image Histogram Information			
5.8.3.8	Signal to Noise Ratio Calculations			
5.8.4	Administrative Messages			
5.9	Format of Integer, Logical and Real Values in Interpretation Data			
5.9 5.9.1				
	in Interpretation Data			
5.9.1	in Interpretation Data Integer Type Data			
5.9.1 5.9.2	in Interpretation Data Integer Type Data Logical Data			
5.9.1 5.9.2	in Interpretation Data Integer Type Data Logical Data			
5.9.1 5.9.2 5.9.3	in Interpretation Data Integer Type Data Logical Data Real Type Data			
5.9.1 5.9.2 5.9.3	in Interpretation Data Integer Type Data Logical Data Real Type Data METEOSAT Dissemination Schedule			
5.9.1 5.9.2 5.9.3 6.	in Interpretation Data Integer Type Data Logical Data Real Type Data METEOSAT Dissemination Schedule Key to the METEOSAT Dissemination Schedule			
5.9.1 5.9.2 5.9.3 6.	in Interpretation Data Integer Type Data Logical Data Real Type Data METEOSAT Dissemination Schedule Key to the METEOSAT Dissemination Schedule			
5.9.1 5.9.2 5.9.3 6. 6.1 6.2	in Interpretation Data Integer Type Data Logical Data Real Type Data METEOSAT Dissemination Schedule Key to the METEOSAT Dissemination Schedule Timing of High-Resolution-Disseminations.			
5.9.1 5.9.2 5.9.3 6. 6.1 6.2	<pre>in Interpretation Data Integer Type Data Logical Data Real Type Data METEOSAT Dissemination Schedule Key to the METEOSAT Dissemination Schedule Timing of High-Resolution-Disseminations.</pre> Technical Specifications of PDUS equipment			
5.9.1 5.9.2 5.9.3 6. 6.1 6.2	<pre>in Interpretation Data Integer Type Data Logical Data Real Type Data METEOSAT Dissemination Schedule Key to the METEOSAT Dissemination Schedule Timing of High-Resolution-Disseminations. Technical Specifications of PDUS equipment Ground Station</pre>			
5.9.1 5.9.2 5.9.3 6. 6.1 6.2 7. 7.1	<pre>in Interpretation Data Integer Type Data Logical Data Real Type Data METEOSAT Dissemination Schedule Key to the METEOSAT Dissemination Schedule Timing of High-Resolution-Disseminations. Technical Specifications of PDUS equipment Ground Station Image Handling System</pre>			
5.9.1 5.9.2 5.9.3 6. 6.1 6.2 7. 7.1 7.2 7.3	<pre>in Interpretation Data Integer Type Data Logical Data Real Type Data METEOSAT Dissemination Schedule Key to the METEOSAT Dissemination Schedule Timing of High-Resolution-Disseminations. Technical Specifications of PDUS equipment Ground Station Image Handling System System Description `</pre>			

METEOSATSYSTEM GUIDERef : M.S.9.0.1.1Volume 9Section 0Issue 4Valid from 01.09.1984

7.4.2 PDUS Front-end Specification

7.4.3 Modulation Specifications

LIST 'OF FIGURES AND TABLES

Fig.	1	METEOSAT Dissemination Mission
Fig.	2	Example of METEOSAT-VIS-Image
Fig.	3	Example of METEOSAT-IR-Image
Fig.	4	Example of METEOSAT-WV-Image
Fig.	5	Example of GOES Image
Fig.	6	High Resolution Formats A and B
Fig.	7	Example of A-Format (Visible)
Fig.	, 8	Example of A-Format (Infrared)
Fig.	9	Example of A-Format (Water Vapour)
Fig.	10	Example of B-Format (Visible)
Fig.	11	Example of B-Format (Infrared)
Fig.	12	Example of a B-Format in polar stereographic projection (visible)
Fig.	13	$\hbox{\tt High Resolution Format X} \qquad \bullet$
Fig.	14	Example of X-Format (Infrared) .
Fig.	15	Examples of BI, BV and BW-Formats
Fig.	16	Examples of BIW, BIVW, and BIV-Formats
Fig.	17	Examples of AI, AV, and AW-Formats
Fig.	18	Examples of AIVH, AIVW, AIW and AIV-Formats
Fig.	19	Coordinate Systems 🕻 50
Fig.	20	Position of the Earth during Scanning
Fig.	21	Coordinate System for Description of Image Deformations
Fig.	22	Max. Deformation Differences
Fig.	23	Image Conditions (CONDS)
Fig.	24	Example of the METEOSAT Dissemination Schedule
Fig.	25	Block Diagram of a PDUS

INTRODUCTION

This document is intended to provide users planning to receive and exploit METEOSAT High Resolution Disseminations with the technical and operational specifications necessary to design and operate a Primary Data User Station (PDUS). It is assumed that the reader has already studied the document:

"Introduction to the METEOSAT System"

Further information and scheduling notices can be obtained from:

European Space Operations Centre Meteosat Exploitation Project (MEP) Att. R. Wolf Robert-Bosch-Str. 5

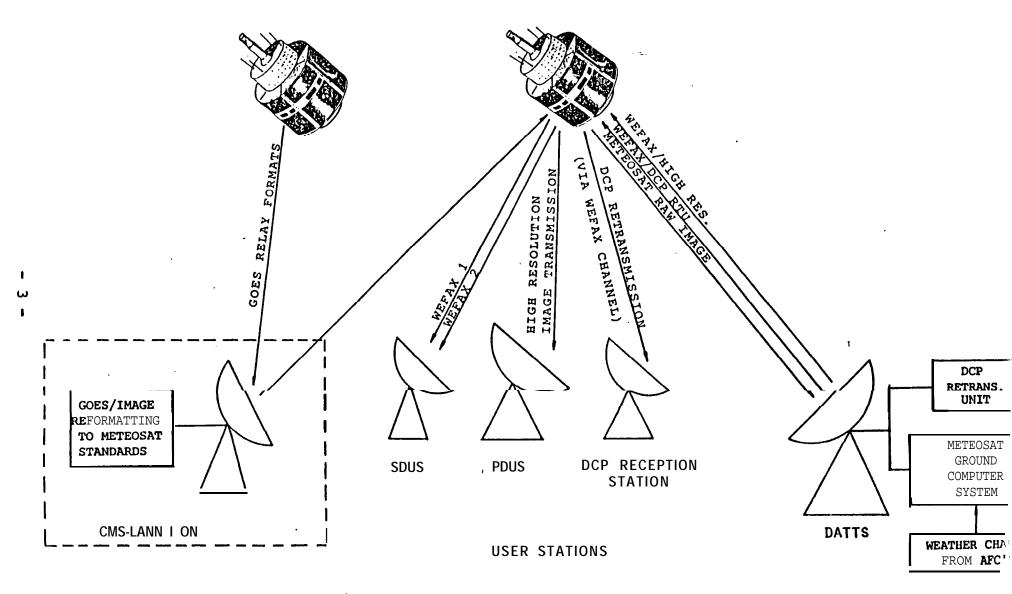
D-6100 Darmstadt

1. DISSEMINATION OF IMAGES

METEOSAT dissemination is the process whereby image data and other meteorological information are relayed via the, spacecraft to the user community.

The spacecraft has two dedicated dissemination channels operating in the S-Band at 1691.0 MHz and 1694.5 MHz, which are both used to distribute a wide variety of data. Two forms of transmissions are employed: conventional analogue transmissions (WEFAX), and special digital transmissions (HR=High Resolution). The WEFAX transmissions are dedicated to user stations for picture recording and use the APT (Automatic Picture Transmission) format. Information about these METEOSAT transmissions can be obtained from the document "METEOSAT WEFAX Transmissions":

The HR digital data are in a format specific to METEOSAT and are designed for users requiring full resolution data in a form suitable for local computer processing.



FIG, 1 METEOSAT DISSEMINATION MISSION

METEOSAT High Resolution Transmissions can be received by "Primary Data User Stations (PDUS)" which, at the simplest form, consist of:

Issue 4

- parabolic antenna
- low noise amplifier
- down converter
- receiver with demodulator
- bit- and frame-synchroniser
- image processor

The actual configuration of the PDUS is related to the special requirements of the user. Some of the possible features of a PDUS are:

- . high resolution hard copy unit
- colour display
- ${\scriptstyle \bullet}$ computer for processing of meteorological data

The images distributed via HR-Dissemination have the full resolution of the raw-image. Therefore the following options could be performed using a PDUS:

- display of images adding false colours or other image enhancements
- computer enlargements of image segments
- . production of animation loops generated from METEOSAT image areas

- remapping of images into different projections
 (e.g., Mercator, Polar Stereographic)
- production of hard copies of the
 - canplete image
 - special areas of the image
- retransmission of special formats via landlines to other stations (e.g. in APT format)
- merging the image data with other inputs, such as
 - radar data
 - conventional meteorological data
 - forecasts or analyses

In addition, certain meteorological products could be derived from the high resolution data such as:

- cloud motion vectors
- sea surface temperatures
- cloud analysis

However, because of the complicity **of** the processing required (and the high resulting cost) it is not recommended to develop such systems but to use the derived meteorological products as produced by the Meteosat Exploitation Project at ESOC, Darmstadt, FRG.

Ref: M.S.9.0.1.1 Valid from 01.09.1984

It is impossible to list all options for a PDUS-design and hence also impossible to quote the costs of a total station. These costs are affected strongly by the extent of an image handling system included in the station.

However, several manufacturers produce such stations and information can be obtained directly from them. For a list of manufacturers **see** document:

"Manufacturers and Suppliers of METEOSAT User Stations and Data Collection Platforms". Volume 9 Section 0 Issue 4 Valid from 01.09.1984

· 3. IMAGE SOURCES

The images disseminated as high-resolution formats are from two different sources: METEOSAT and GOES-E.

3.1. METEOSAT

METEOSAT takes pictures of three spectral bands:

Visible ($0.4 - 1.1 \mu m$) Infrared (10.5 - 12.5 μm) Water Vapour ($5.7 - 7.1 \mu m$)

The infrared and water vapour images are composed of 2500 pixels and 2500 lines with a spatial resolution of 5 km at the sub-satellite point.

The visible picture contains 5000 pixels and **5000** lines in case both visible channels should be simultaneously used. The resulting resolution is 2.5 km at the sub-satellite point.

Assuming nominal image scanning is performed, every half hour one set of pictures is available. This set could consist of

- either 2.5 km resolution visible and S-km infrared or
- 5 km resolution visible, infrared and water vapour

Section 0

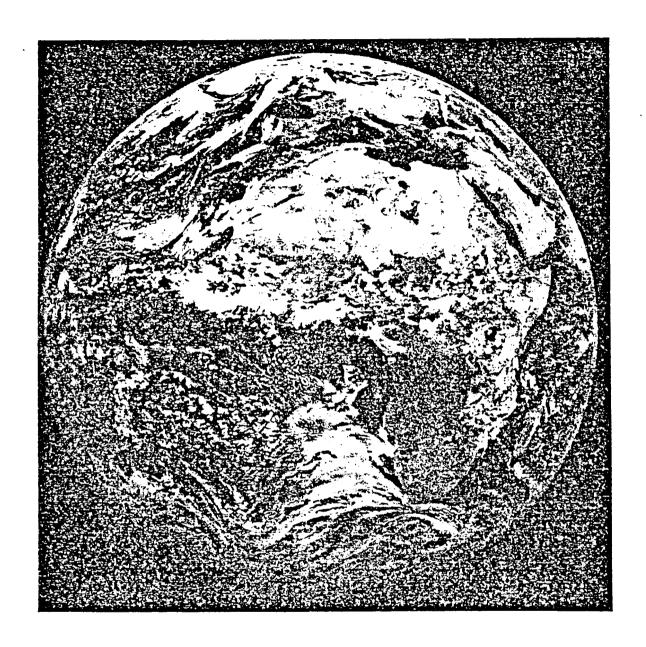


FIGURE 2 Example of METEOSAT-VIS-Image

In this full disk image taken by visible light, clouds appear white and space is seen as black.

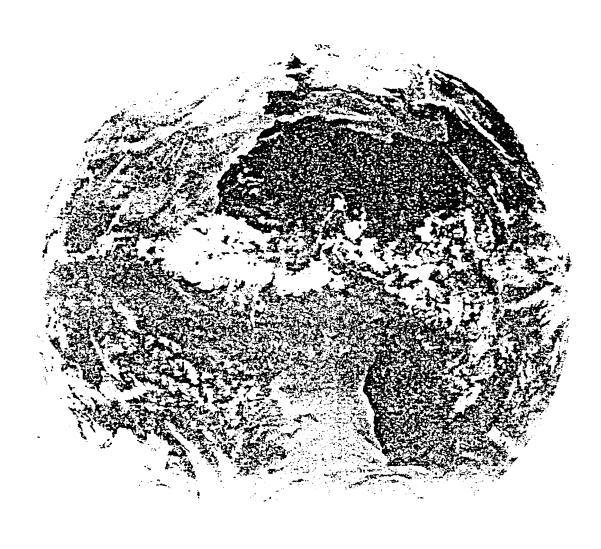
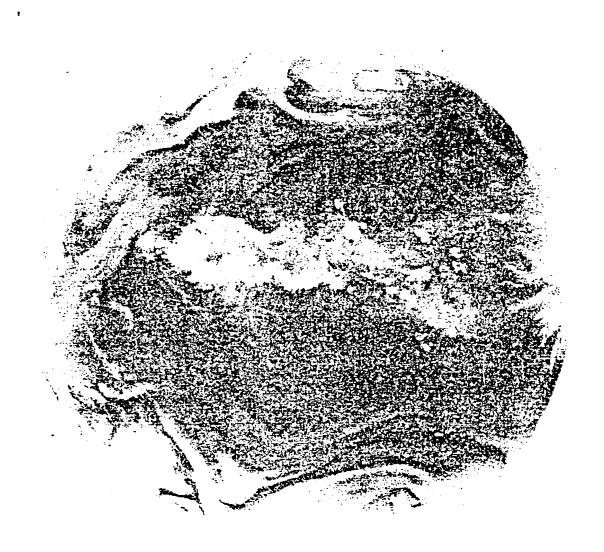


Figure 3 Example of METEOSAT-IR-Image

This image utilises the 11 µm "window" band of infra-red radiation, in which grey scales represent temperature. Space and cold high clouds appear white, warmer (low) clouds grey, and hot deserts as very dark areas.

Issue 4

Valid from 01.09.1984



The channel used is the 6 µm water vapour absorption band of infra-red radiation. White areas represent low temperatures, and this can be equated with areas of high humidity. The darker shades indicate a lower level of atmospheric

Example of METEOSAT-WV-Image

FIGURE 4

humidity. The image represents the upper atmosphere only and does **not** provide an earth view

Future METEOSAT spacecrafts will be able to deliver **VIS1,** VIS2, IR and WV information simultaneously. The first launch of this type of spacecraft **is** planned for 1987 (MOP 1).

The visible and water vapour images are coded in 64 grey-levels, the infrared image in 256 levels. After pre-processing the 64 grey-levels of the visible and water vapour image are expanded to 256 levels.

Future METEOSAT spacecrafts will deliver all information in 8 bits corresponding to 256 levels.

3.2. GOES-E

The GOES-E satellite located in a geostationary orbit at 75°W observes in two spectral bands:

visible channel : 0.55 - 0.75 pminfrared channel : $10.5 - 12.6 \text{ }\mu\text{m}$

In the infrared the "Visible and Infra-Red Spin/Scan Radiometer" (VISSR) scans one line on each spin, providing roughly 7 km resolution at the sub-satellite point. In the visible the VISSR scans eight parailel lines on each spin designed to provide approximately 1 km resolution at the sub-satellite point. Visible channel scan lines are combined in groups of two, four or eight to provide 2, 4 or 7 km resolution respectively.

Ref : M.S.9.0.1.1

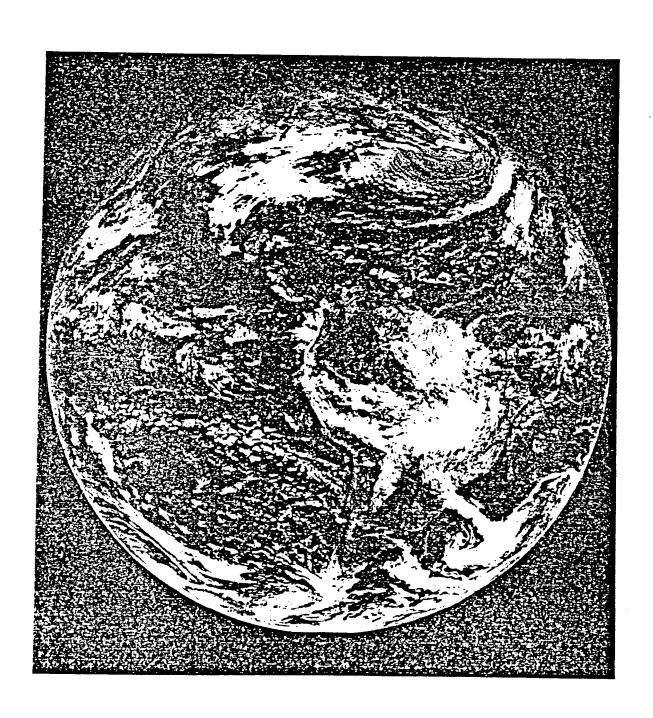
Valid from **01.09.1984**

Section 0 Volume 9

Issue 4

A full disc image. consists of 14 000 lines in the visible spectrum and 1750 lines in the infrared It is obtained in 17.5 minutes. Full disc pictures are normally scheduled at half hourly intervals, but for special occasions (for example during the hurricane season) the operation is changed to include more frequent scans of smaller areas.

Volume 9 Section 0 Issue 4 Valid from 01.09.1984



Ref : M.S.9.0.1.1 Valid from **01.09.1984**

4. HIGH-RESOLUTION FORMATS

METEOSAT High-Resolution Formats are mainly originated from ESOC Darmstadt. Only the relay transmission of the American GOES-E satellite located over Central-America is formatted in the Centre Météorologique Spatiale in Lannion (France). Both Centres transmit the formats in the same mode via satellite. The transmission speed is 166.66 kbit/sec.

4.1 High Resolution Formats originated in ESOC

All raw-image data as received at the ground computing centre from the satellite are normally "pre-processed", i.e., an amplitude processing in such a way is performed that lines still correspond to physical scans. Registration errors caused by the differing fields of view of the sensors are corrected.

The pre-processed image is subject to a geometrical correction in order to canpensate the shift deformation caused by the spacecraft's deviation from its nominal geostationary position. The scanned image is "moved" (rectified) into the reference image frame which corresponds to the image as seen from the nominal. satellite position.

The geometrical image processing is based on an image geometry model which derives its inputs from horizon extraction on the actual infrared image and results in a very accurate fine attitude determination. The rectification of the current image is calculated from

Ref: M.S.9.0.1.1 Valid from 01.09.1984

the deformation vector field obtained from the image geometry model of the current image. This image geometry model is updated dynamically and describes the movement of an image related reference frame as function, of radiometer scan step. This allows to calculate the deformation vector for every image point on the earth. All results of the image processing chain are stored in a cyclic file. This file is designed to cover sufficient history (e.g., 100 days) in order to perform long term quality and trend analysis. The described rectification of images allows the uncomplicated production of loop sequences of images.

4.1.1. A-Format

The format A represents the whole earth-disc as seen by METEOSAT's visible, infrared or water vapour sensors. The image information of one, two or three sensors can be transmitted in one format. The following lists the total range of possible combinations.

Format	VIS	IR	WV	Transmission Time (min)
AV	2500 lines VIS1 * 5000 pixels + 2500 lines VIS2 * 5000 pixels or 2 *2500 lines VISX * 5000 pixels			23.5
AI		2500 lines * 2500 pixels	-	б
AW			2500 lines * 2500 pixels	6
AIV	as AV	as AI		29.3
AIW		as AI	as AW	. 11.8
AIVW	2500 lines VIS* * 5000 pixels	a8 AI	a8 AW	23.5
AIVH	2500 lines VISX * 2500 pixels Data reduction by taking only each second pixel within the line	as AI		11.8

* VISx - VIS1 or VIS2 depending on selected sensor channel

Bef : **M.S.9.0.1.1**Issue 4 Valid from **01.09.1984**

With the exception of format AIVH, all images have the same resolution as the original METEOSAT raw-image. An A-Format consists of 5000 pixels and 5000 lines in the visible spectrum and 2500 pixels and 2500 lines in the infrared and water vapour spectra. The sub-satellite paint resolution therefore is 2.5km in the visible and 5km in the infrared and water vapour spectrum. In AIVH formats the visible information is reduced to 2500 pixels and 2500 lines by suppression of every second pixel within an Image line and using the information of one visible channel only.

Ref : M.S.9.0.1.i
Valid from 01.09.1984

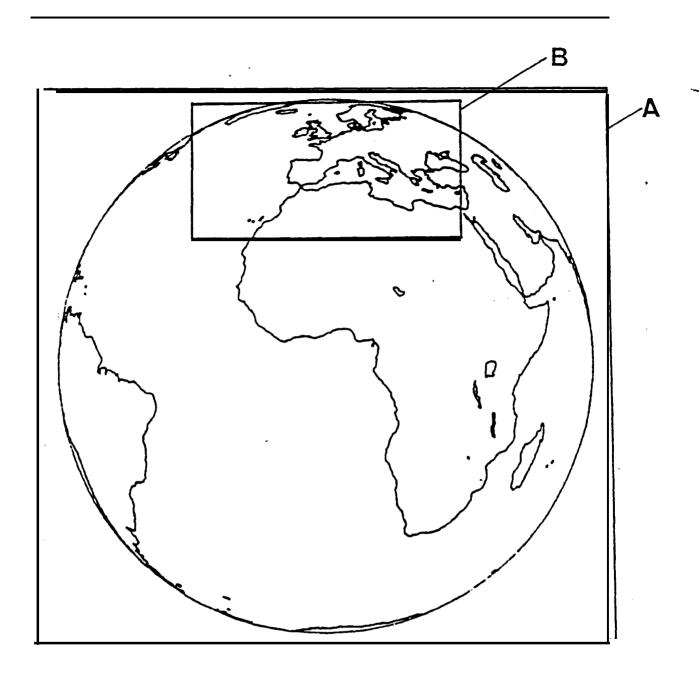


FIGURE 6 High-Resolution Formats A and B disseminated from ESOC, Darmstadt (METEOSAT images)

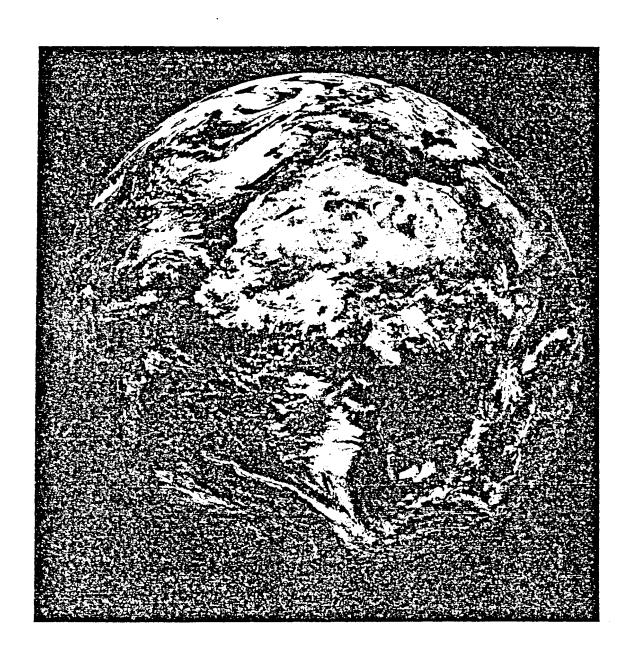


FIGURE 7 Example of A-Format (Visible)

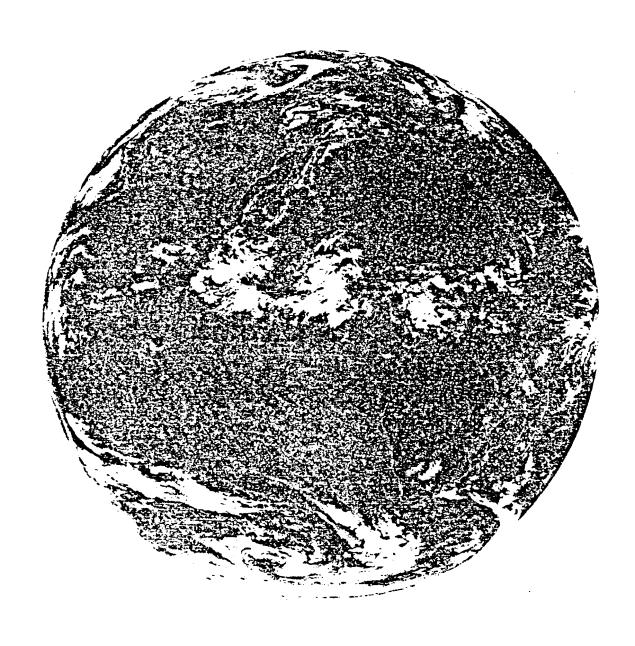


FIGURE 8 Example of A-Format (Infrared)

Ref : M.S.9.0.1.1 Volume 9 Section 0 Issue 4 Valid from **01.09.1984**

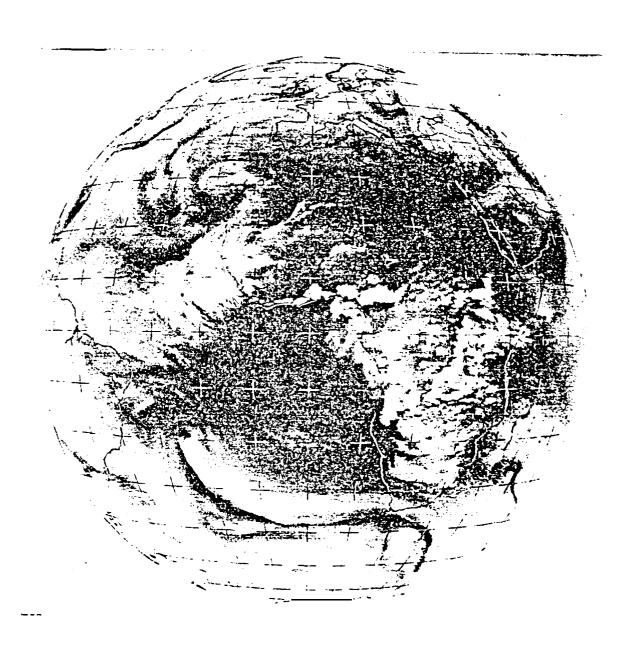


FIGURE 9 Example of A-Format (Water Vapour)

E Ref: M.S.9.0.1.1 Issue 4 Valid from 01.09.1984

4.1.2. B-Format

Format P represents the European, North-African and Middle-Eastern regions as seen by METEOSAT's visible, infrared and water vapour sensor. The transmission of a B-Format could contain the information of o.e, two or three spectra. The following formats could be produced:

Format	VIS	IR	WV	Tranamlaaion Time (min)
BV	625 lines VIS1 + 625 lines VIS2 * 2500 pixels or 2 * 625 lines VISX * 2500 pixels			3.1
BI		625 lines * 1250 pixels		0.9
BW			625 lines * 1250 pixels	0.9
BIV	as BV	as BI		3.8
BIW		as BI	as BW	1.6
BIVW	625 lines VISX * 2500 pixels	as BI	as BW	3.1

* VISx = VIS1 or VIS2 depending on selected sensor channel
The first line number of B-formats is 1810, the last
line 2434, the first pixel within the lines la 626, the
last 1875 in case of IR and WV information. The first
pixel within VIS lines is either 1252 or 3751.

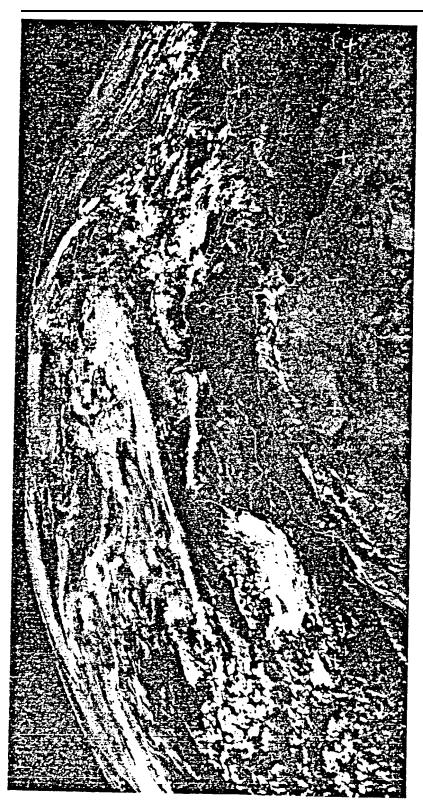
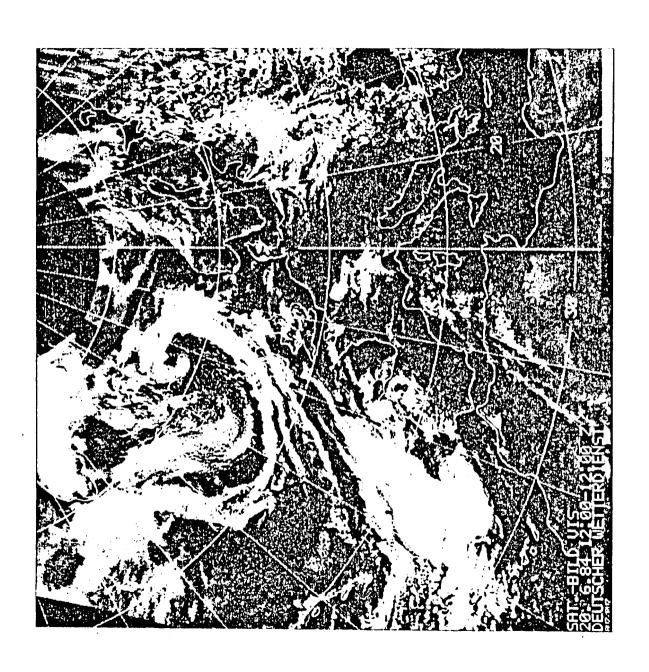


FIGURE 10 Example of B-Format (Visible)

Volume.9 Section 0 Issue 4 Valid from 01.09.1984



FIGURE 11 Example of B-Format (Infrared)



B-Format (visible) projection FIGURE 12

Ref: M.S. 9.0.1.1 Valid from 01.09.1984

4.2. High-Resolution Formats originated in CMS-Lannion

4.2.1. X-Format

The X-Format represents the North and South American continents as seen by the visible and infrared sensors on board the American GOES-E satellite. Visible and infrared images are transmitted simultaneously. The X-Format consists of 2500 pixels and 2500 lines in the visible and 1250 pixels and 1250 lines in the infrared spectrum. The resolution therefore is 7 km at the sub-satellite point in the infrared and 2 km in the visible band.

The transmission time of the X-Format is 7.4 minutes.

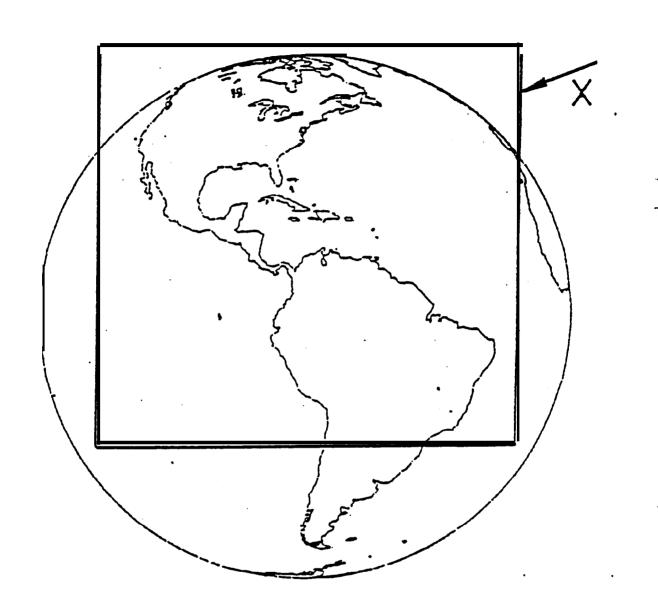


FIGURE 13 High Resolution Format X disseminated from CMS-Lannion (GOES-E image)

Volume 9 Section 0 Issue 4 Valid from 01.09.1984

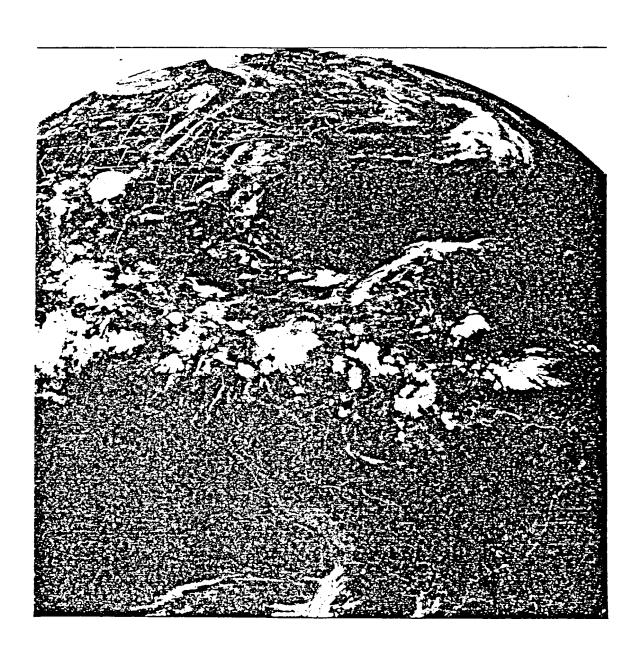


FIGURE 14 Example of X-Format (Infrared)

Section 0

Issue 4

Ref: M.S.9.0.1.1 Valid from 01.09.1984

5. THE DIGITAL TRANSMISSION FORMAT

5.1. Definition of Terms

Formats are structured into Subframes and Frames.

FORMAT:

Data-Set used for the transmission of image information.

A-Formats : AV, AI, AIVW...

B-Formats : BV, BI, BIVW . . .

X-Formats : XI, XV, XIV . . .

PRAMES:

Each Data Frame consists of 364 words of 8 bits.

SUBFRAMES:

Subframes consist of 4 or 8 frames as follows:

A-Format-Subframes = 8 Frames

B- **or** X-Format Subframes = 4 Frames

DEFINITION FRAME / SUBFRAME:

1.	364 words of 8 bits	1 Frame
2.	-	B-Format
3.		Subframe
4.		
5.		
6.		A-Format Subframe
7.		dollane
8.		

Volume 9 Section 0 Issue 4 Valid from 01.09.1984

5.2. Format Construction

A complete Format consists of the following sets of Subframes:

- 1. Heading Subframes
- 2. Image Data and Grid Information Subframes
- 3. Annotation Subframes *
- 4. Conclusion Subframes

FORMAT CONSTRUCTION

HEADING SUE	BFRAME
DATA SUBFRAMES	•
ANNOTATION	*
CONCLUSION	

* B-Formats only

Valid from 01.09.1984

5.3. Elements of Subframes

Before the corstruction of Subframes is explained In detail It is necessary to describe Identification, synchronisation, interpretation and other data fields used in the Subframes.

These data are:

Synchronisatlon Word

ID-Word

Label

Identification

Interpretation Data

Fillers: SP1, SP2, SP4, SP5, and SP6

Grid Information

Synchronisation Word 5.3.1.

The Synchronisation Word is identical in all Subframes. It consists of 3 bytes as follows:

(MSB) 00000i0i 0000ii00 11011111 (LSB)

■ Most Significant Bit MSB

LSB ■ Least Significant Bit

5.3.2. ID-Word

The ID-Word consists of 1 byte (8 bits). The value of the ID-Word is increased by 1 for each Frame within a Subframe.

The first ID-Word of the first Frame in each A-Format Subframe is:

(MSB) 01110000

Volume 9 Section 0 Issue 4

Pef : M.S.9.0.1.1
Valid from 01.09.1984

Due to the fact that the A-Format Subframes consist of 8 Frames, the ID-Word of the last Frame in each A-Format Subframe Is:

(MSB) 01110111

The ID-Word of the first Frame in each B-Format and X-Format **Subframe** is:

(MSB) 00110000

Due to the fact that B- and X-Format Subframes consist of 4 Frames, the ID-Word of the last Frame In each B- or X-Format **Subframe** is:

(MSB) 00110011

By interpretation of the ID-Word it is easy to distinguish between A- and B- or X-Formats. The difference between both is in bit 2 of the <code>ID-Word</code> which is:

1 for A-Formats

and

0 for B- and X-Formats

5.3.3. Label

The contents of the Label are as follows:

Bytes

- 1 + 2 Number of Frames in the Subframe:
 - 8 for A-Formats
 - 4 for B- and X-Formats

Ref: M.S.9.0.1.1

Section 0 Issue 4 Valid from 01.09.1984

Bytes

- 3 + 4 Total number of Subframes in the Format counting all header Subframes as 1 and all conclusion Subframes as 1. (The total number of Subframes therefore is 2 plus the amount of Data and Annotation Subframes).
- Current Subframe Number

 The current Subframe Number for all Header
 Subframes and all Conclusion Subframes = 1.

 The counting of Data Subframes starts with for the first Data Subframe. The first
 Annotation Subframe has the current Subframe
 number 625.
- 7 + 8 Image Line Number

 The Image Line Number is Ø in the Header,

 Annotation and Conclusion Subframes.

 In A- and X-Formats the first Image Line Number

 1, in %-Formats the first Image Line Number

 1810. In all cases, the Image Line Number

 increases by 1 for each set of Subframes.
- 9 + 10 Image Number from start of the Mission 11 + 12
 - 13 Format Indicator (hexadecimal)
 - $A = \emptyset \emptyset$
 - B = FF
 - $X = \emptyset F$

01.09.1984

Valid from

Bvtes

Volume 9

14, 15, Indicators of spectral information in the

16, 17 Subf rame

Byte 14 = VIS 1

Byte 15 = **VIS** 2

Byte 16 = IR

Byte 17 = WV

These indicators are set to the following hexadecimal values:

ØØ if corresponding data are not represented in the Subf rame

FF If corresponding data are represented in the Subf rame

Due to the fact that subframes containing full resolution VIS Information contain only half lines bytes 14 and/or 15 are used In Data Subframes as follows:

FØif 1st half of VIS line is present in the Data Subframe

ØF if 2nd half of **VIS** line is present in the Data Subframe

In Data Frames containing reduced VIS information (AIVH-Format) bytes 14 or 15 are set to FF.

NOTE Header Subframe Bytes 14 and 15 Indicate only that spectral information is present In the following transmission. Bytes 14 and/or 15 can be FF or ØØ:

Ref : M.S.9.0.1.1
Issue 4 Valid from 01.09.1984

Volume 9 Secti

Section 0 Issue

Bytes

- 18 Indicator of Grid Information:
 - ♥ if no Grid Information is present
 - ϕ F if reference Grid Information is present
- 19 Indicator of Annotation Text:
 - FF if Annotation Text is present
 - $\phi \phi$ if Annotation Text is not present
- 20 Indication of Scan Direction:
 - ØØ if Scan is S-N and E-W
 - $\mathbf{F}\phi$ if Scan is N-S and E-W
 - ØF if Scan is S-N and W-E
 - FF if Scan is N-S and W-E

Normally Byte 20 has the hexadecimal value $\varphi \varphi$ for METEOSAT (A- and B-Formats) and FF for GOES Belay Information (X-Format)

- 21 + 22 Spare bytes
- 23 + 24 All bytes are set to $\phi\phi$

5.3.4. Identification

Bytes

1 + 2 Satellite Indicator

The Satellite Indicator is used to give information about the image source of the transmitted format:

GOES = $\phi\phi\phi\phi$ (hexadecimal)

METEOSAT 1 = D4E3 = MT (EBCDIC - CODE)

METEOSAT 2 = D4F2 = M2 (")

 $METEOSAT_n = D4F_n = Mn (")$

Bytes

Volume 9

- 3 + 4Year of Image Acquisition This information is coded in binary values, e.g., 1982.
- 5 + 6Day of Image Acquisition This information is coded in binary values, e.g., 346 for the 346th day of the year.
- 7 + 8Nominal Image Time This information is coded in binary (BCD -Binary Coded Decimal) and indicates Hour and Minute of completion of Image Acquisition, e.g., 0000 - 2330.
- 9 +10) As for the
- 11+12) Label
- in Header Subframes 13-20)
- 21-32 Spare bytes all set to $\varphi \varphi$

5.3.5. Interpretation Data

The Interpretation Data Block consists of 1360 bytes and is designed for information on satellite orbit, geographical correction methods used, calibration coefficients and-administrative messages.

Section 0

Issue 4

Valid from 01.09.1984

5.3.6.Fillers

Fillers were included to the Data Transmission in order to keep Subframes at a fixed structure. The fillers are as follows:

SP1 = **8** bytes, all set to $\phi\phi$

SP2 = 16 bytes, all set to $\phi\phi$

SP3 = Not in use

SP4 = 1440 spare bytes (4x360), all set

to $\phi\phi$

SP5 = 40 bytes, all set to $\phi\phi$

SP6 = 8 bytes, all set to $\phi\phi$

5.3.7. Grid Information

Grid Information is enclosed in each Data Subframe. The n^{th} bit of the Grid Information corresponds to the n^{th} byte of the Image Data of the same Data Subframe. The bit is set to 1 if grid point is available.

The Grid Information of A-Formats consists of 316 bytes = 2528 bits.

The first 2500 bits are the Grid Information corresponding to 2500 image pixels within the Subframe. 28 spare bits are set to \emptyset .

The Grid Information of B-Formats consists of 158 bytes = 1264 bits.

The first 1250 bits represent the Grid Information to the 1250 image pixels transmitted within the Subframe, 14 spare bits are set to \emptyset .

Volume 9 Section 0 Issue 4

X-Formats do not contain separate Grid Information. Grids and Coastlines are superimposed to the images. The Grid Information field (158 bytes) is set to ϕ .

Construction of Subframes: 5.4.

5.4.1. Heading Subframes

5.4.1.1.Heading Subframes of A-Formats

The Heading Subframes of A-Formats consist of 8 frames of 364 bytes each. The structure is as follows:

----- 364 bytes -----

SYNC	ID	LABEL	SPl	IDENTIFI	CATION	SP2	INTERPRET.DATA (280 bytes)
SYNC	ID	INTERPR	ETATI	ON DATA	(360	bytes)	
SYNC	ID	INTERP	RETATI	ON DATA	(360	bytes)	
SYNC	ID	INTERPF	RETATI	ON DATA	(360	bytes)	
SYNC	ID	SP4					
SYNC	ID	SP4					
SYNC	ID	SP4					
SYNC	ID	SP4					

The Heading Subframe is transmitted 42 times for each A-Format.

Section 0 Volume 9 Issue 4

5.4.1.2. Heading Subframes of B- and X-Formats

The Heading Subframes of B- and X-Formats consist of 4 frames of 364 bytes each. The ettucture is as follows:

----- 364 bytes -----

SYNC	ID	LABEL SP1 IDENTIFICATION SP2	INTERPRET.DATA (280 bytes)
SYNC	ID	INTERPRETATION DATA (360 bytes)	
SYNC	ID	INTERPRETATION DATA (360 bytes)	
SYNC	ID	INTERPRETATION DATA (360 bytes)	

The Heading Subframe is transmitted 84 times for B-Formats and 90 times for X-Formats.

Volume 9

Section 0 Issue 4 valid from 01.09.1984

Bef : M.S.9.0.1.1

5.4.2.Data Subfraree

5.4.2.1. A-Format Data Subframes

A-Format Data Subframes consist of 8 frames of 364 bytes each. The structure is as follows:

----- 364 bytes -----

SYNC	ID	LABEL SE	25 DA	TA (2	296 byte	es)
SYNC	ID	DATA	· ·	(360	bytes)	
SYNC	ID	DATA		(360	bytes)	
SYNC	ID	DATA		(360	bytes)	
SYNC	ID	DATA		(360	bytes)	
SYNC	ID	DATA		(360	bytes)	
SYNC	ID	DATA		(360	bytes)	
SYNC	ID	DATA (44	bytes)	(316	bytes)	GRID

Each A-Format Data Subframe includes 2500 bytes of Image Information. This corresponds to 2500 Image pixels and represents either one complete line of an IR or WV Image or half a line of one VIS channel image (different for AIVH formats).

Valid from 01.09.1984

5.4.2.2.B- and X-Format Data Subframes

B- and X-Format Data Subframes consist of 4 frames of 364 byte6 each. The structure is as follows:

SYNC	ID	LABEL	SP6	DATA (328 bytes)
SYNC	ID	DATA	(360	bytes)
SYNC	ID	DATA	(360	bytes) . *
SYNC	ID	DATA	(202	bytes) (158 bytes) GRID

364 bytes -----

Each B-Format Data Subframe include6 1250 bytes of Image information. This corresponds to 1250 image pixels and represents one complete B-Format line of ${\bf 1R}$ or WV image or half a B-Format line of one VIS Channel image.

5.4.3. Annotation Subframes

These-frames are no longer transmitted.

Volume 9 Section 0 Issue 4

5.4.4. Conclusion Subfrares

The Conclusion Subframe is a copy of the Heading Subframe.

The amount of Conclusion Subframes transmitted in the individual Transmission Format 1s:

for all A-Formats One

* * * *

for BI, BIV, BW, BVW AND X-Format6

Two for BIW, BIVW and BV.

Valid from 01.09.1984

5.5. Sets of Data Subframes

A set of Data Subframes represents one Image Line of all Spectral Channels included in the individual Transmission Format.

A set of Data Subframes therefore consists of one or a combination of the following elements:

One Subframe of IR Data Two Subframes of VIS2 Data Two Subframes of VIS1 Data

One Subframe of WV Data

Consequently, the number of Data Subframes in a set of Subframes is varying in accordance with the number of Spectral Channel Information included in the Transmission Format. The formats include the following information:

AI, BI, XI	Subframe	IR only
AIV, BIV, XIV	Subframes	IR + VIS2 + VIS1
AIW, BIW	Subframes	IR + WV
AIVW, BIVW	Subframes	'IR + WV + VIS (2 or 1)
AW, BW	Subframe	WV only
AV, BV	Subframes	VIS only (VIS1 + VIS2
AIVH	Subframes	Reduced VIS (2 or 1) + IR

Within the sets of Subframes the Data Subframes of the different Spectral Information are transmitted in the sequence:

IR, VIS2, VIS1 (or WV)

Ref : M.S.9.0.1.1
Valid from 01.09.1984

5.6. Transmission Formats

A list of available Transmission Formats is given in Chapters 4.1.1. (A-Formats), 4.1.2. (B-Formats) and 4.2.1. (X-Formats). For better understanding, the construction of some Transmission Formats is shown in Figures 15, 16, 17 and 18.

In case that information of one Visible Channel is not available, the information of the other Visible Channel is repeated in Formats AV, BV, AIV and BIV.

HEADING SUBFRAMES	84 SF 1	HEADING SUBFRAMES	845F	HEADING SUBFAMMES	84 SF 236 F
IR LINE 1840 IR LINE 1841 IR LINE 1842 IR LINE 1844 IR LINE 1845 IR LINE 1847 IR LINE 1847 IR LINE 1848 IR LINE 2929 IR LINE 2930 IR LINE 2931	625 Subfrakes = 2500 Francs	VIS 2 1/2 LINE ARAO VIS 2 1/2 LINE ARAO VIS 3 1/2 LINE ARAO VIS 2 1/2 LINE ARAA VIS 2 1/2 LINE ARAA VIS 3 1/2 LINE ARAA VIS 1 1/2 LINE ARAA VIS 1 1/2 LINE ARAA VIS 2 1/4 LINE ARAA VIS 3 1/4 LINE ARAA	2500 SUBFRAHES = 10 000 FRAHES	WV LINE ASAO WV LINE ASAA WV LINE ASAA	625 SUBFRAHES = 2500 FRAMES
IR LINE 2432 IR LINE 2433 IR LINE 2433		VIS 2 1/2 LINE 2434 VIS 1 1/2 LINE 2434 VIS 4 1/2 LINE 2434		WV LINE 2432 WV LINE 2433 WV LINE 2434	
ANNETATION	30SF 1.00F 1.00F	ANNOTATION	30SF 170F	ANNOTATION	305F
CONCLUSION	135	CONCLUSION	25. 8.	CONCLUSION	\$ s

BI BV BW

 $\underline{ t FIGURE 15}$ Examples of BI, BV and BW-Formats

HEADING SUBFRAMES	84 SF = 336 F	HEADING SUBFRAHES	845F = 336 F	HEADING SUBFRAMES	845F= 336 F
IR LINE 1810 WY LINE 1811 IR LINE 1811 IR LINE 1812 WY LINE 1812 IR LINE 1813 WY LINE 1813	SUBFRAMES = FRAMES	1R LINE 1810 VISZ 1/2 LINE 1810 VISZ 1/3 LINE 1810 WV LINE 1810 IR LINE 1811 VISZ 1/2 LINE 1811 VISZ 1/2 LINE 1811 WV LINE 1811	SUBFRAMES = FRAMES	IR LINE 1810 VIS2 LINE 1810 VIS2 1/2 LINE 1810 VIS1 1/2 LINE 1810 VIS1 1/2 LINE 1810 IR LINE 1811 VIS2 1/2 LINE 1811	%3 ¢
IR LINE 1814 IR LINE 2432 1 UV LINE 2432	1250 SUBFRA 5000 FRAMES	IR LINE 1812 VK2 1/2 LINE 2433 WV. LINE 2433	1500 SUBFRAN NO 000 FRANES	VISA 1/2 LINE ARM VISA 1/2 LINE 2433 IR LINE 2434	3125 SUBFRAMES 12500 FRAMES
IR LINE 2433 WY LINE 2433 IR LINE 2434 WY LINE 2434 ANNOTATION		IR LINE 2434 VIS2 1/2 LINE 2434 VIS2 1/3 LINE 2434 WV LINE 2434 ANNOTATION	305F. - 120F.	VIS 2 1/2 LINE 2434 VIS 2 1/2 LINE 2434 VIS 1 1/2 LINE 2434 VIS 1 1/2 LINE 2434 ANNOTHTION	305
CONCLUSION	25. 8F	CONCLUSION	255	CONCLUSION	2SF 3

* IF MUNOTATION SIT IS SET ONLY . A CONCLUSION SUBTRAME

BIW BIVW BIV

FIGURE 16 Examples of Formats, BIW, BIVW, and BIV

HEADING SUBFRAMES	42 SF. 336 FRAUS	HEADING SUBFRAMES	42 SF= 336 FRAUES	HEADING SUBFRAMES	42 SF = 336 FRMUS
IR LINE A IR LINE 2 IR LINE 3 IR LINE 4 IR LINE 5 IR LINE 6 IR LINE 7 IR LINE 8 IR LINE 9	SUBFARHES = FRANES	VK2 1/2 LINE 1 VB1 1/2 LINE 1 VB1 1/2 LINE 1 VB1 1/2 LINE 2 VB1 1/2 LINE 3	SUBFARMES = FRANES	WV LINE A WV LINE 3 WV LINE 4 WV LINE 5 WV LINE 6 WV LINE 7 WV LINE 8 WV LINE 9	SUBFRAMES =
IR LINE 2495 IR LINE 2497 IR LINE 2497 IR LINE 2499 IR LINE 2499 IR LINE 2499	2500	VIST JLING 2499 VIST JLING 2499 VIST JLING 2500	000 08	WV LINE 2495 WV LINE 2496 WV LINE 2497 WV LINE 2499 WV LINE 2499 WV LINE 2500	2500
CONCLUSION	A. 7	CONCLUSION	X. 4	CONCLUSION	7. 3

FIGURE 17 Examples of AI, AV, and AW-Formats

Valid from 01.09.1984 : M.S.9.0.1.1

3. 3.	NOISATINOS
ш е	0257 2N178 FSIA
	OOST JAITY, 1811
	0058 3N174251A
	0057 20174 2511
100	OQSZ SNIT YI
2 500	65AZ 9N174, VSA
SUBFRAMES FRAMES	Z 3N177, FSIA
ગ ફૂ	2 34177/2511
Ä	7 3N174, 751A
4	7 3N17 YI
	L 20174 1511
	1 2m173,154
	y 30174, 75M
	V 3M73/ 25/
	V JNI VI
336	534441005
35 2	
	IN/AU3H

AIV

7.7	NOISATONOS
	MY LINE 2500
	OSZ JAM YI
	PPY LINE 2499
	6312 2N17 YI
	85AZ SNT AM
5000	86AZ SNIT VI
5000	i
SUBFRAMES FRAMES	3N17 AM
HES =	4 2017 81
	E 3NIT AM
	8 ANIT 11
	2 2NIT AM
	1 3N17 XI
	. 3N/7 AM
	V 3N17 81
334	SAHBYIONS
42 SF .	9N/AV3H
	AIW

2" "	NOISATONOO	2" 2	NOISATTAOT
SO DOD SUBFRAMES =	0057 2N17 AM 0057 2N17 SIA 0057 2N17 SIA 0057 2N17 SIA 0057 2N17 YI 66A7 9N17 AM 55A7 9N17 SIA 2 3N17 AM 2 2N17 SIA 2 3N17 SIA 2 3N17 SIA 2 3N17 SIA	SUBFANES =	0052 2N17 HSIA 0052 2N17 HSIA 0052 2N17 HSIA 65A2 2N17 HSIA 66A2 2N17 HSIA 66A2 3N17 HSIA 5 3N17 YI 4 2N17 YI 5 3N17 HSIA 5 3N17 HSIA
42 SF *	V 3N17 101 V 3N17, ZSIA	336 FAMIS	2 3M17 HSM 2 V 1M2 V NSM V 3M17 HSM V 3M17 V SAUBUSH MEUDING
	AIVW		AIVH

48

FIGURE 18

Examples

0f

AIVH,

AIVW,

AIW and AIV-Formats

Ref : H.S.9.0.1.1

5.7. File Layout of Interpretation Data

The "Interpretation Data" black contains 1360 bytes and organised as follows:

4.	Administrative Messages	(800	bytes)
3.	Imagery Data	(328	bytes)
2.	Spacecraft Operations Data	(128	bytes)
1.	Calibration Data	(104	bytes)

5.7.1. Calibration Data

OFFSET	NAME	TYPE	LENGTH	CONTENT
0	BBC	ASCII	5	Normalized Black Body Count
5	SDBBC	ASCII	3	xxx.xx Standard deviation for BBC
				x.xx
8	TIME1	ASCII	5	DDDSS (Day -D, Slot -S)
				Timestamp for BBC
13	IRCAL	ASCII	5	IR-Calibration value 0.0XXXXX
18	TIME2	ASCII	3	Timestamp for IR-Calibration
				value (DDD)
21	WVCAL	ASCII	4	Water vapour calibration value
				o.ooxxxx
25	TIME3	ASCII	3	Timestamp for WV-Calibration
				value (DDD)
28	GAINS	AS CII	8	Gains for IR, WV, VIS1 & VIS2
				G1G1GwGwGvGvG2G2
36			68	Spare for future use

Section 0 Ieeue 4 Valid from **01.09.1984**

5.7.2. Spacecraft Operation Data

OFFSET	NAME	TYPE	LENGTH	CONTENT
104	DEGSRA	R*8	8	Right ascension attitude
				south in degrees
112	DEGSDE	R*8	8	Declination attitude south
				in degrees
120	DECNRA	R*8	8	Right ascension attitude
				north in degrees
128	DEGNDE	R*8	8	Declination attitude north
				in degrees
136	FINATT	R*8	24	X, Y, Z of refined attitude
160	FARADE	R*8	16	Right ascension and
				declination of mean refined
				attitude •
176	NRSLOT	I *4	4	Number of slots for refined
				attitude
18.0	SPNDUR	R*4	4	Spin duration minus nominal
in the second of a	CH.			spin duration in microseconds
184	FLECL	L*1	1	Eclipse operation
185	PLDEC	L*1	1	Decontamination
186	FLHAN	L*1	1	Manaeuvre
187	FLMODE	L*1	1	TRUE-earth, FALSE-sun '
188	FLIR1	L*1	1	IR1 on
189	FLVISI	L*1	1	VIS1 an
190	FLVIS2	L*1	1	VIS2 on
191	FLWV	L*1	1	WV on
192	FLIR2	L*1	1	IR2 on
193			39	Spare for future use

Volume 9 Section 0 Issue 4 Valid from 01.09.1984

Ref: M.S.9.0.1.1

5.7.3. Imagery Data

OFFSET	NAME	TYPE	LENGTH	CONTENT
232	IMSTAT	L*1	16	<pre>1 Horizon analysis 2 Spin speed fit 3 Orbit offset vector fit 4 Pixel sampling rate fit 5 Attitude refinement iteration based on horizon results 6 Not in use 7 Not in use 8 Calculation of deformation vector field 9 Not in use 10 Completion of rectification 11 Canpletion of amplitude processing 12-</pre>
248	LIMHOR	I*2	24	16 Spare Southern line, 1. pixel/last pixel Northern line, 1. pixel/last pixel East pixel, south line/north line West pixel, south line/north line

OFFSET	NAME	TYPE	LENGTH	CONTENT
272	SATDIS	R*8	8	Distance satellite - earth
				centre
280	SORBOF	R*8	24	x, y, z components of orbit
				offset
				vector in nominal image frame
				at southern horizon limit
304	NORBOF	R*8	24	x, y, z components as above at
				northern horizon limit
328	XDDIFM	R*4	4	Max. deformation difference
				x-canponent within column
332	YDDIFM	R*4	4	Max. deformation difference
				y-canponent within lines
336	XSCM	R*4	4	Max. deformation difference
				x-canponent within lines
340	YSCM	R*4	4	Max. deformation difference
				y-canponent within columns
344	CONDS	L*1	4	1. Earth within field of view
				in east/west
				2. Earth within field of view
				in south/north ·
				3. East/west horizon limits
				within predicted margin
				4. South/north horizon limits
				within predicted margin
348	LOWDYN	I"2	10	Lowest non-zero count in
				histogram (-1 if off) for IR1,
				VIS1, VIS2, WV and IR2

Ref : M.S.9.0.1.1

Volume 9 Section 0 Issue 4 Valid from 01.09.1984

ŀ					
	OFFSET	NAME	TYPE	LENGTH	CONTENT 1
	358	HIGDYN	If2	10	<pre>Highest non-zero count in histogram (-1 if off) for IR1 VIS1, VIS2, WV and IR2</pre>
	368	MVISl	R*4	4	Mean vaiue of VISl
	372	MVIS2	R*4	4	Mean value of VIS2
	376	SNNOM	R*4	12	IR1/IR2, VIS1/WV, VIS2
					nominal reference count
					signal to noise ratios
	388	SNNLIN	I*4	4	Number of lines used for S/N
					calculation
	392	SNREP	R*4	12	S/N ratios eastern part for
					<pre>IR1/IR2, VIS1/WV and VIS2</pre>
	404	SNRWP	R*4	12	S/N ratios western part for
					IR1/IR2, VIS1/WV and VIS2
	416	SWMNEP	R*4	12	Mean signal value eastern
					part for IR1/IR2, VIS1/WV and
					VIs2
	420	SIMMWP	R*4	12	Mean signal value western
					part for IR1/IR2, VIS1/WV and
					vIS2
	440	SNMXEP	I*2	6	Maximum space count eastern
					part for IR1/IR2, VIS1/WV and
					VIs2
	446	SMMXWP	I*2	6	Maximum space count western
					part for IR1/IR2, VIS1/WV and
					vIS2
	452				108 spare bytes for future
					use'

5.7.4. Administrative Messages

OFFSET NAME TYPE LENGTH CONTENT

560 ADMNS ASCII 800 Administrative Messages

- 5.8. Content of Interpretation Data
- 5.8.1. Calibration Data
- 5.8.1-1. Normalised Black Body Count (BBC)

The Black Body Count (BBC) to be transmitted is that normalised to a temperature of 290K. Variations in the normalised BBC can be considered to provide information on variations in response of the radiometer and can thus be used to fine tune the MIEC IR calibration value. This is done by calculating the FAG-Factor (Fine Adjustment of Gain) from the normalised BBC. The reference value for the nonnalised BBC has been chosen as 110 and the FAG Factor is defined as 110/BBC. The corresponding radiance value for a measured IR count is given by:

R =(Count - 6) \times CAL \times (110/BBC) where CAL is the MIEC IR calibration value. Temperatures can be derived using a table relating Radiances to Planck Temperatures.

Section 0

Volume 9

Issue 4

Ref : M.S.9.0.1.1 Valid from 01.09.1984

The figure to be sent in the interpretation data is normalised BBC represented by five ASCII figures which have to be read as follows:

XXX.XX

Only the five X figures are transmitted where the last two figures are the decimals of the normalised BBC.

5.8.1.2. Standard Deviation for BBC (SDBBC)

This value represents the difference between the last two BBC measurements and gives an indication on the present development on the radiometer response.

Three ASCII characters are transmitted. The first represents the total, the two others the-decimals of the value (X.XX).

5.8.1.3. Timestamp for BBC (TIME1)

This timestamp represents the day and slot of the last BBC measurement. The timestamp is transmitted as 5 · ASCII characters in the following layout:

DDDSS

DDD Day of the year

SS = Slot of the day

Ref: M.S.9.0.1.1 Valid from 01.09.1984

5.8.1.4. IR-Calibration value (IRCAL)

The IR-Calibration value is that used in MIEC (Meteorological Information Extracting Centre at ESCC) processing to convert from IR Counts to radiances. It is updated as necessary as a result of canparisons between sea surface temperatures derived from METEOSAT image data and those from conventional ship measurements. The five ASCII characters transmitted should be interpreted as follows:

O.OXXXXX

5.8.1.5. Timestamp for IR Calibration Value (TIME2)

This timestamp represents the day of the year (DDD) when the IR-Calibration value was introduced.

5.8.1.6. Water Vapour Calibration Value (WVCAL)

The WV-Calibration value is that used in MIECprocessing to convert from WV counts to radiances. It
is updated as necessary as a result of comparisons
between radiosonde data and Upper Tropospheric Humidity
data calculated from METEOSAT image data. The four
ASCII characters transmitted should be interpreted as
follows:

0.00XXXX

Valid from 01.09.1984

5.8.1.7. Times&up for WV-Calibration Value (TIME3)

This timestamp represents the day of the year (DDD) when the WV-Calibration value was-introduced.

5.8.1.8. Gains (GAINS)

The actually selected gains of all spectral bands are transmitted in 8 ASCII characters with the following layout:

GiGiGwGwGvGvG2G2

GiGi = Gain IR GwGw = Gain WV

GvGv = Gain VIS1

G2G2 = Gain VIS2

The gain represents an amplification factor of the radiometer electronics. The gain can be selected by telecommand between 00 and 15.

The transmission of two blanks in one of the gain patterns indicates that this spectral channel is switched off.

The purpose of the gain change is to correct for variations in the radiometer response so that the maximum voltage variation in the sensor output analogue signal corresponds to the maximum radiance variation experienced by the radiometer.

Issue 4

In nominal operations, the gain settings are chosen in a way that the individual detector is only marginally saturated at the warmest (IR) or brightest (VIS) image slot of the day. However, for special projects, a gain setting could be selected to enhance lower radiances, if interest is only in these. This would result in saturation of the higher radiances.

5.8.2. Spacecraft Operations Data

This data set describes the satellite position in space. The coordinate frame used for this description is called \$\coldon 50\$ and is related to the spring equinox in the year 1950. The coordinate frame is an intertial frame with its origin in the Earth's centre. The X-axis points in the direction of mean vernal equinox. The Z-axis points in the direction of celestial north (the' star Polaris) and the Y-axis forms the 3rd member of the trihedron. Figure 19 shows the coordinate frame \$\forms\$ 50.

Ref: M.S.9.0.1.1
Valid from 01.09.1984

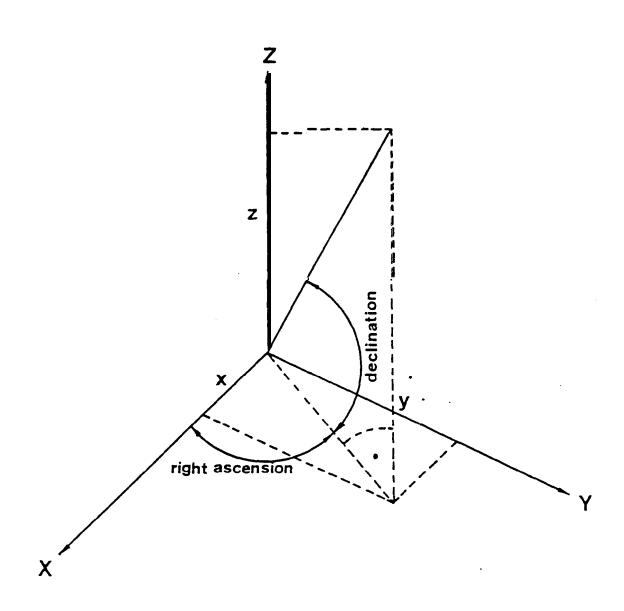


FIGURE 19 Coordinate System § 50 (Spring Equinox in 1950)

5.8.2.1 Right ascension attitude south in degrees (DEGSRA)

This value' describes the angle between the X-axis and the vector formed by the projection of the attitude vector (momentary axis of rotation) projected onto the XY level. This value is measured during the scanning of the southern image part.

5.8.2.2 Declination attitude south in degrees (DEGSDE)

This value describes the angle between the attitude vector and its projection on the XY level. This value is measured during scanning of the southern image part.

5.8.2.3 Right ascension attitude north.in degrees (DEGNRA)

This is the value corresponding to 5.8.2.1 but measured during scanning of the northern image part.

5.8.2.4 Declination attitude north in degrees (DEGNDE)

This is the value corresponding to 5.8.2.2 but. measured during scanning of the northern image part.

Ref: M.S.9.0.1.1 Valid from 01.09.1984

5.8.2.5 X, Y, Z of refined attitude (FINATT)

These are the values of projected attitude vector \mathbf{X} , \mathbf{Y} , and \mathbf{Z} canponents (see Fig. 19, values \mathbf{X} , \mathbf{Y} , \mathbf{Z}) based on the refined attitude. This is a mean of attitude vectors over nominally 48 slots.

5.8.2.6. Right ascension and declination of mean refined attitude (FARADE)

This is a mean of the last (nominally 48) values of the right ascension and declination.

5.8.2.7. Number of slots for refined attitude (NRSLOT)

This is the amount of slots (nominally 48) used for calculation of the refined attitude. This-amount of slots was used for calculation of values 5.8.2.5. and 5.8.2.6.

5.8.2.8. Spin duration (SPNDUR)

This time is measured during scanning of line 1250 nominally (middle of the image) corresponding to nominal radiometer position 1350. The value transmitted is the measured spin duration in usec minus the nominal spin duration (600 000 µsec).

If the satellite rotation is faster than nominal the value will be negative.

5.8.2.9. Flags for special operation modes

The various operation modes are indicated by a L*1 value. If the value is set 1 the operations mode was active during acquisition of the image. The values are:

FLECL - Eclipse Operation

FLDEC - Decontamination of the radiometer is operated

FIMAN - S/C manoeuvre was performed

FLMODE - Satellite is in Earth mode if value is 1

in Sun mode if value is \$\infty\$

FLIR1 - IR1 switched on

FLVIS1 - VIS1 on

FLVIS2 - VIS2 on

FLWV - WV on

FLIR2 - IR2 on

5.8.3. Imagery Data

These data are produced during image processing in the MGCS-MF. They are intended to inform the user on the amount of processing which was necessary to move the original raw image into a reference frame. In addition, information on amplitude analysis will be provided. The data are:

Image and processing status information
Radiometer step offset at image start
Satellite position (distance, attitude, orbit)
Conditions resulting from horizon analysis
Amplitude analysis (dynamic ranges)

Issue 4

Ref : M.S. 3. U. 1. 1 Valid from 01.09.1984

Image and processing status information (IMSTAT)

These data are represented by 16 L*1 values which are set 1 (.TRUE.) if the corresponding item was successfully completed (ϕ = .FALSE.). The 16 bytes are dedicated as follows:

- 1 Horizon analysis
- 2 Spin speed fit
- 3 Orbit offset vector fit
- 4 Pixel sampling rate fit
- 5 Attitude refinement iteration based on horizon results
- 6 Not in use
- 7 Not in use
- 8 Calculation of deformationvector field
- 9 Not in use
- 10 Campletion of rectification
- Completion of amplitude 11 processing
- 12 16 Spare

5.8.3.2. Southern line etc (LIMHOR)

The groups of values in LIMHOR indicates the position of the Earth as seen during scanning. The 12 values are (see Fig. 20):

Southern line:. The number of the first 1 scanning linecontaining Earth information

information 3 Last pixel: As above, but the column of the last detected Earth pixel in the line 4 Northern Line: The number of the last scanning line containing Earth information 5 1. pixel: The column of the first detected pixel in the last line containing Earth information 6 Last pixel: As above, but the column of the last detected Earth pixel in
the last detected Earth pixel in the line 4 Northern Line: The number of the last scanning line containing Earth information 5 1. pixel : The column of the first detected pixel in the last line containing Earth information 6 Last pixel : As above, but the column of the
In the line 4 Northern Line: The number of the last scanning line containing Earth information 5 1. pixel : The column of the first detected pixel in the last line containing Earth information 6 Last pixel : As above, but the column of the
scanning line containing Earth information 5 1. pixel : The column of the first detected pixel in the last line containing Earth information 6 Last pixel : As above, but the column of the
information 5 1. pixel : The column of the first detected pixel in the last line containing Earth information 6 Last pixel : As above, but the column of the
detected pixel in the last line containing Earth information 6 Last pixel : As above, but the column of the
line containing Earth information 6 Last pixel : As above, but the column of the
information 6 Last pixel : As above, but the column of the
6 Last pixel : As above, but the column of the
-
the line .
7 East pixel : The right most pixel (column
number) inthe image
a South line : Number of the first scanned
line containing the right most
pixel
9 North line : Number of the last scanned line
containing the right most pixel
10 West pixel : The left most pixel (Column number) inthe image
11 South line · : Number of the first scanned line
containing the left most pixel
12 North line : Number of the last scanned line
containing the left most pixel

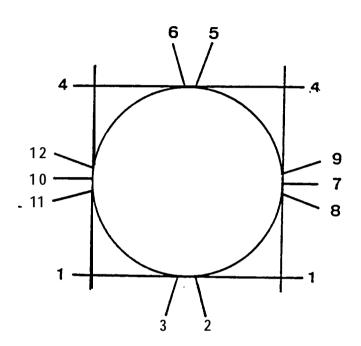


FIGURE 20 Position of the Earth during Scanning (LIMHOR)

Issue 4

Ref: M.S.9.0.1.1 Valid from 01.09.1984

5.8.3.3. Distance Satellite-Earth Centre (SATDIS)

This double precision value indicates the distance of the satellite to the Earth centre in km at the time of scanning the Earth centre.

5.8.3.4. X, Y, Z Components (SORBOF, NORBOF)

These values indicate the level of changes to the image during image geometry corrections. For the description of the image geometry the following set of coordinate systems is used:

- The nominal image frame (NIF) with origin at nominal satellite position N, Xn-axis pointing towards the earth centre and Zn=0 being the equatorial plane.
- The actual image frame (AIF) with origin at momentary satellite position S, Xa-axis pointing towards the earth, Za-axis coinciding with attitude vector a and the earth centre always being in the plane Ya = 0. Generally this frame is moving with each scan step of the radiometer.

(In the subsequent description the subscripts "a" and "n" refer to **AIF and'** NIF respectively).

The principal idea of the image geometry model is then to adjust the actual image frame at position of southern respectively northern horizon limit scans and to **express its** movement (offset vector and Euler angles) with respect to the nominal image frame as functions of the radiometer scan step. Conservation

Ref : M.S.9.0.1.1 Valid from 01.09.1984

of angular momentum of the spacecraft (except for manoeuvres) is used for determining an approximate mean attitude vector as starting point for an iterative attitude refinement procedure per image.

Since infrared radiation even at the **polar** caps of the earth differs at all times significantly from thermal noise corresponding to space the southern and northern horizon scan limits can well be extracted from the infrared image.

In order to get these positions accurate to part of a pixel a parabola arc is fitted to the horizons. The radiation threshold value defining the horizon and the apparent radius of the earth are adjustable parameters in order to compensate for the thickness of the atmosphere.

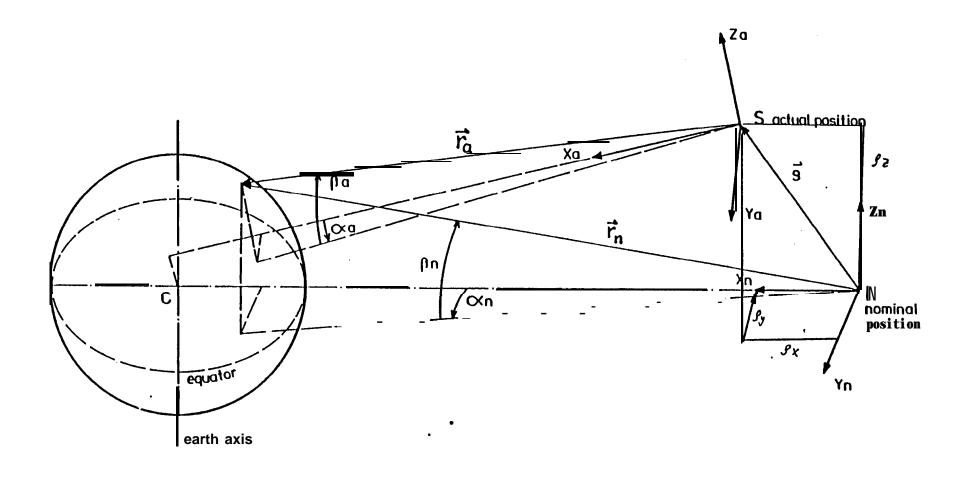


Figure 21 Coordinate system for description of image deformations.

The earth center C is always in the planes $y_a = 0$ and $y_n = 0$.

Figure 21 Coordinate systems for description of image deformations. The earth center C is always in the planes $y_a = 0$ and $y_n = 0$.

5.8.3.6. Image Conditions (CONDS)

These logical values are set .TRUE. when Earth was within the field of view during scanning. (1 for E/W, 2 for N/S). 3 and 4 are set .TRUE. when the scanned Earth image was within ± 3 pixels from the predicted values (see Figure-23).

5.8.3.7. Image histogram information (LOWDYN, HIGDYN, MVIS1, MVIS2)

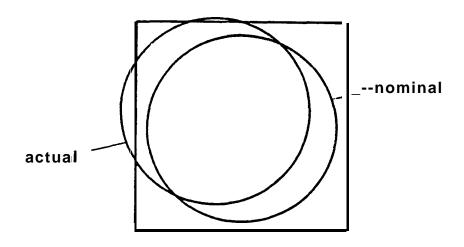
These values indicate the dynamic of the individual spectral channels.

LOWDYN indicates the lowest count in the histogram for IR1, VIS1, VIS2, WV, and IR2 (in this order). If the channel is switched off the value -1 is transmitted. HIGDYN is the highest count (contrary to LOWDYN).

MVIS1 and MVIS2 represent the mean value of VIS1 and VIS2 image pixels in space (measurement for noise in space).

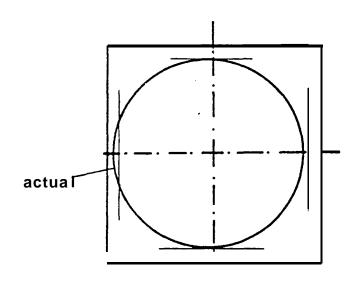
All these informations could be used for image enhancement in order to make use of the full dynamic range of hard copy recorders or display units. Section 0

Issue 4 Valid from **01.09.1984**



CONDS : Both . FALSE.

1,2:0



CONDS :1:1

2:1

3:0 4:1

FIGURE 23 Image Conditions (CONDS)

5.8.3.8. Signal-To-Noise Ratio Calculations (SNNOM, SNNLIN, SNREP, SNRWP, SWMNEP, SNMNWP, SNMXEP, SNMXWP)

These parameters give information on the quality of the . radiometer during the -scanning.

For each channel the pixel values in space to either side of the Earth are examined to determine the noise level in this image part. The value is calculated as follows:

P NS = SQRT (SUM (J**2*K(J)/F) J=1

S

where F= SUM K(J) J=1

 ${f J}$ represents the pixel value and K(J) is the number of pixels with value ${f J}$

snnom contains the nominal reference count signal to noise ratios of IR1/IR2, VIS1/WV, VIS2

SNNLIN contains the amount of lines used for the above described S/N calculation

SNREP & give the S/N ratios for western and
SNRWP eastern part

Ref : M.S.9.0.1.1

Valid from 01.09.1984

SWMNEP & give the mean signal value of western and SNMNWP eastern part

Issue 4

SNMXEP & give the maximum space count of western SNMXVP and eastern part

5.8.4. Administrative Messages

Up to 800 ASCII characters can be used for administrative messages. These are used to inform users about changes in operations of the METEOSAT system. The messages are marked with an "Administrative Message Number" which consists of the year and month of issue and a sequence number of messages issued in this month. Volume 9 Section 0 Issue 4 Valid from 01.09.1984

Format of Integer, Logical and Real Values in 5.9 Interpretation Data

5.9.1. Integer type data (I*2, I*4)

Two-byte and four-byte integer types are represented as 16-bit and 32-bit binary numbers, respectively. first bit indicates the sign. A positive integer is represented by a 0 in the sign bit and a binary true number representation in the remaining bits. A two's canplement with 1 in the sign bit represents a negative integer. All bits contain Os to represent zero.

5.9.2. Logical data (L*1)

One-byte and four-byte logical data are represented as 8-bit and 32-bit binary numbers, respectively. value has 1 in the final bit with all remaining bits 0. A false value has **Os** in all bits.

FORTRAN 77 determines that a logical data value is true when any bit is 1 (true), and that the value is false when all bits are 0 (false).

5.9.3. Real type data (R*4)

Real type data is represented as a 32-bit floating point number. The first bit indicates the sign. bits following the sign bit indicate the exponent part, the remaining 24 bits constitute a six hexadecimal digit mantissa with each of the four bits representing a digit.

Ref : **M.S.9.0.1.1**Issue 4 Valid from **01.09.1984**

The hexadecimal radix point **of** the mantissa is to the left of the most significant mantissa digit. The exponent part represents a number in the range of 0 to 127. **It** represents an exponent in the range of -64 to 63, after subtraction of 64 from the actual exponent.

0		8	16	24	31
s	E			M	

5.9.4. Double precision data (R*8)

Double precision data is represented as a 64-bit floating point number. Its sign and exponent part are similar to those in real data. The mantissa consists of 56 bits following the exponent part and has 14 hexadecimal digits (each four bits represents a digit).

0	8	8	16	24	32	40	48	56	63
S	E				М				

- **S**: eign (positive if 0 and negative if 1)
- E: exponent. This value is obtained by adding $\mathbf{2^6}$ to the actual exponent.
- M : hexadecimal mantissa part with the radix point to the left of bit 8.

Value = Man. 16 exp.

Ref : **M.S.9.0.1.1**Valid from 01.09.1984

6. **METEOSAT** DISSEMINATION SCHEDULE

The spacecraft has two dedicated **dissemination** channels which can be **used** for both analogue and digital transmission. At the preaent **time** the following philosophy **is** used to determine the METEOSAT **dissemination** schedule:

- 1) Channel 1 1694.5 MHz
 Analogue transmissions from ESOC-Darmstadt
- 2) Channel 2 1691.0 MHz All digital data (A, B and X formats transmitted to PDUS), plus all transmissions made from CMS Lannion (identified by L in the schedule) plus transmissions- of processed data (CTH, weather Charts).

• In addition, administrative message6 are transmitted

simultaneously on both channels in WEFAX format, and

space is reserved on both channels simultaneously for

the operation of METEOSAT Ranging measurements.

This concept is subject to modification at short notice for a variety of operational considerations, such as:

- reduced computer capacity
- availability of only one dissemination channel
- special eclipse operational mode
- maintenance considerations

Schedule

the METEOSAT Dissemination

Example of

24

FIGURE

SOME FORMATS ARE SUBJECT TO . DUE TO SEASONAL UNDERJ' 'MINATION

.09.1984

6

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HHNN 26194192269343342466594659	B1 B3 B4 D5 B6 B7 De B2 D9	47 48 49 48 48 48 48 40.	BI	1 (((((((((((((((((((1 6 1 6 1 6 1 6 1 6 1	D 2 D 3 D 4 D 5 D 7 D 8 D 9 D 1 D 3	6. 6. 6. 6. 7. 7.	A I D T Q I		E3 1 D2 2 C82 C83 D1 D3 1 D4 D5 D6 D2 C82 C83 C3D	12. 12. 12. 12. 12. 12. 12. 13. 13.	BIV DTO AI LR ETO BIV	2 V12 T12 1 2 12 T12 13	CH1 P2 C 9 2 C 9 3 D 1 D 3 D 4 D 5 D 6 D 7 C 1 D	18 18 18 18 18 19 19 19	BIV	2 1 1 1 1 1 1 1 1 1	CHI E3 D2 tCe2 C03 B1 D3 D4 B5 D6 D2 C92 C93 C1D	22: 24. 24. 24. 24. 24. 25. 25. 25.	CH2 BIV LX BIV LX AI	I II /42/ 24! 24! 24! 24! 25! 25!	CH 4 L D 2 C 0 2 C 0 3 D 1 D 3 D 4 D 5 D 6 D 2 C 0 3 C 1 D	30 30 30 36 30 30 31 31 31	AI DTOT ETOT BIV AV	1 391 391 1(391 391 311 311 1[CH1 E3 D2 D1 D3 D6 D7 BB D7 BB D1 D1	34: 36. 36. 36. 36. 36. 37. 37. 37.	BIVUDTOTAL Al LR 3 UEFA BIVU	361 361 361 361 610 51 1 371 361 371	D2 D1 D3 D4 D5 D7 D8 D2 D9 4	42. 42. 42. 42. 42. 42. 43. 3. 43.	BIU AI TOTO	1 421 421 21 431 431	2 6 14 18 22 26 38 34 38 42 46
2 6 1 1 1 4 1 8 22 26 3 1 1 3 4 3 8 4 2 4 6	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	2 2 2 2 3 3 3 3		14 1	3	P2 DI D3 E1 E2 E3 E4 (ES ID2 D1 D3 E6	8 8 8 8 8 8 9 9 9 9	BIV	911		14. 14. 14. 14. 14. 14. 15. 15. 15.	BIV	14 14 H14 	(I I C 3 D I C 4 B I C 9 2 I C 9 3 I C 6 D 6 7 I B 9 2 1 . I C 9 2 I C 9 3 I C 9 3	19. 19. 20. 20. 20. 20. 20. 20. 20. 20. 20. 20	BIV	20 20 420 1 21	C3D C02 C02 C03 C5D C6D D7 De D9 2 C03 C7D	25. 25. 26. 26. 26. 26. 26. 27. 27.	LZ CH2 TVH2 DTOT AI CSB CSB CSB	I I I 24 I 6 I B 26 I 26 I 26 I 26 I 27 I 27 I 27 I 27 I	CH 1 CJD C4B 2 3 C82 C63 C5D 6D 3: D7 D9 3 2 3 C62 C63 C7D	31. 31. 32. 32. 32. 32. 33. 33. 33. 33.	BIVU At U 3	32 32 2 33 33	D2 3: D1 3 E 1 E 2 E 2 E 5 D B 1 D B 3 E 1 E 5 D B 1 D B 3 E 1 D B	3.8.38.38.38.38.38.38.38.38.38.38.38.38.	I AM 3 U AM 3 U AM 3 I AM 3	381 381 371 1 371 1 910	CH1 . 2 44. D1 D3 2 45. D1 D3	BI 144. 44. 44.	CH2 4 A1 AU	4 1 1 4411 4411 4511	22 26 39 34 38 2 46 SD
2 6 10 14 1D 22 26 30 34 38 42 46	I I I ID2 ID3 ITE: IRAI IRAI IRAI ID2 ID3	CHI 4 4 8T E HG 0	TI ai	CH2	41 41 4 5 51	E9 D2 D1 D3 ADHI RAN(D2 D1 D3	10.6 10 H 9. G 0.	BIU AI RANG BIU	19 61 19 19 19 19 19 19 19 19 19 19 19 19 19	I D2 I D2 C02 I C03 I TES I RAH! RAH! I C03 I C03 I TES	16.16.16.16.17.17.17.17.1	. CH . UEF . BIV TEST . RA RANG . BIV 7 .	2 I A 31 V16 61 16 16 17	ICSD LCSD LOS LCOS ICOS IADMI I ICOS LCOS ICOS ICOS ICOS	21 22 22. 22. N 9. G 9. R 3. B 3. A 23.	.BIV ADNI AI .RAH ANG IV	2 t 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	C9D	27 28. 28. 20. 10. 29. 29.	BIVU TEST AL	2 II ! 28! 4! 26! ! 01 01 29! 29!	C9D D2 C63 C63 ADMI	33: 34 34 34 0.	CH2 BIVU ABHH AI RANG BIV	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	E8 E9 D1 D3 TEST RANG D2 D1	39: 49: 49: 41: 41:	BIW TEST AI WEFF RAHI BIW	2 II II 48 I 48 I 40 I 40 I 7 - I 6 0 I 5 0 I 4 I I 4 I I	D2 D1 D3 ADMH RAHO D2 D1 D3 E1	46 46. 46. 10. 47. 47. 47.	CHA BIU ADM AI RAM RAM BIU	461 461 461 461 461 471 471	26 30 34 38 42 46 50

Valid from 01.09.1984

For this reasdn users are advised that it is expedient to provide equipment capable of receiving both METEOSAT dissemination channels, either simultaneously or by switching between channels. To standardise the start times of individual formats, four minute transmission slots have been introduced. Formats start at the nominal time? H + 02, + 06, + 10 etc., with actual transmissions scheduled to start a few seconds after these nominal times. HR formats may use between 1 (BI) and 8 (AIV) slots. If a format is finished before the end of a dissemination slot the time period between end of this image and the start of the next image is not used on dissemination channel 2. Time slots between consecutive WEFAX transmissions on channel 1 are used for retransmission of DCP data.

METEOSAT Dissemination Schedules listing all disseminations for PDUS and SDUS users are produced by ESOC and can be mailed to all interested persons or organisations on request. An example of this schedule is shown in Fig. 24.

Key to the METEOSAT Dissemination Schedule 6.1

Heading The schedule number is generated with the following design: YYMMC/NN

Y Y = Year of introduction

MM = Month of introduction

= Character which indicates the satellite

N N = Number of schedule introduced in this month

The schedule number is important since changes to schedules will be announced by administrative messages. The information "valid from" is important as schedules are normally distributed in advance.

COLUMN HEADINGS - CH1 is the METEOSAT dissemination channel at 1694.5 MHz

CH2 is the METEOSAT dissemination channel at 1691.0 MHz

The time HH is the HOUR during which transmission starts

ROW HEADINGS - MM gives the minute of the hour for
the start of the transmission

TABLE ENTRIES - give the FORMAT

followed by the slot number of the image used. The slot number is effectively the nearest half hour at the end of the image, e.g., image ending near to 1200 UT = slot 24
Image ending near to 1930 UT = slot 39
The actual end time of the image is part of the header in all digital image transmissions

The formats are indicated with the characters used in the format description (3.3 and 3.4). Formats generated at CMS-Lannion are

Section 0 Issue 4

Ref: M.S.9.0.1.1 Valid from 01.09.1984

FORMATS

Volume 9

indicated by an "L". The numbers following the indicating character define the area of the format (as shown in figures). Slots with the entry RANG are allocated to spacecraft ranging. During these slots no dissemination is scheduled.

Examples: AIV 18 = Infrared plus visible image of the globe as seen by METEOSAT. (canpletion time = 0900 GMT) is used.

> LX 24 = Visible and infrared image of X-sector as seen by GOES-E spacecraft. The image is formatted • in CMS-Lannion and is taken from slot 24 (completion time near to 1200 GMT).

6.2 Timing of High-Resolution Disseminations

BR-Formats are transmitted with the following frequencies:

A every hour (in IR)

B every 1/2 hour

X every 6 hours

In case of B-formats all available spectra with meaningful information for the particular slot are That means that format BIW is transmitted during night times, BIV or BIVW are transmitted when illumination in the visible spectrum is sufficient.

Section 0

Issue 4

Ref : M.S.9.0.1.1 Valid from **01.09.1984**

A-Formats are mainly used as AI or AW. Format AV is scheduled once or twice only due to the long transmission time. The interleaved formats AIV and AIVW are not used at the present time. Format AIVH will after a test period be transmitted hourly replacing AI formats (day time only).

7. TECHNICAL SPECIFICATIONS OF PDUS EQUIPMENT

A PDUS is comprised of the following two sub-systems:

- Ground Station
- Image Handling System

7.1 Ground Station

The Ground Station consists of:

- a) Antenna
 - Low Noise Amplifier (Preamplifier)
 - Downconverter
- b) Receiver
 - Demodulator
- c) Bit Synchronizer
 - Frame Synchronizer

7.2. Image Handling System

The Image Handling System varies very much according to the specifications of the user. It consists of:

- Computer
- Store Unit

Possible options are:

- Colour Image Display
- 'High-Resolution Recorder (Hardcopy system)
- Computer with external links for proceesed data and reformatted image information
- Video Recorder

7.3 System Description

The antenna must be suitable for operation in the 1.7 A front-fed parabolic reflector with a 4.5m diameter is recommended. The receiver consists of the low noise front-end, one or more down-converter8 and the demodulator. The selection of the input frequency of the receiver has to be done by the user. Normally a frequency of approximately 70 MHz18 chosen. the fact that HR-Disseminations may be operated in both dissemination channels' two switchable Input frequencies The recommended bandwidth of the have to be foreseen. The demodulator output will be a receiver is 1 MHz. PCM signal with a bit rate of 166.66 kbit/sec.

Bit Synchronizers with bit rates from 100 bps to.5 M bpe and Frame Synchronizers with wordlengths of 8 to 24 bit8 and frame length of 4 to 256 words are offered by different manufacturers as standard equipment. The selection of word length depends on the computer ueed.

Suitable store media of PDUS image handling systems are either digital tape deck8 or Winchesterdisks. A minimum of 80 M byte disc store is recommended.

Ref : M.S.9.0.1.1
Valid from 01.09.1984

- 7.4. Input/Output Specifications
- 7.4.1 Spacecraft output data

RF carrier frequency

- Channel 1 : 1694.50 MHz - Channel 2 : 1691.00 MHz

7.4.2 PDUS front-end specification -

The recommended C/T ratio for a PDUS 10.5 dB/°K.

Depending on the location, however, a relaxation of this specification is possible. To gain this G/T ratio different system concepts are possible. The usage of a high quality antenna system with a diameter of approximately 4.5 m (antenna gain > 33 dB) allows the use of a low cost transistor preamplifier with a noise figure of approximately 200°K. Asmaller antenna system could, on the other hand, be compensated by using a high quality parametric preamplifier with a noise figure \$\infty\$100°K.

Because of the different specifications for a PDUS by different users and the possible locations of the. station, further technical data for the PDUS are not given In this document, but advice can be obtained from the Agency.

A study for a 'low cost PDUS has been produced for ESA by the University of Dundee. A copy of the study can be obtained from the Agency.

85

FIG, 25 PDUS BLOCK DIAGRAM

7.4.3 Modulation Specifications

Modulation PCM (SP-L)/PM

Index 1.2 rad ± 5%

Bit rate 166.66 kbit/sec

RF/F 660 Hz

min. signal to

noise ratio

C/No for

receiver -67.0 dB.Hz

METEOSAT HIGH RESOLUTION IMAGE DISSEMINATION

October 1989

~ 14 m

Meteosat Exploitation Project

European Space Operations Centre Robert-Busch-Strasse 5 6100 Darmsvadt Federal Republic of Germany

Table of Contents

1.0	INTRODUCTION	
2.0	PRIMARY DATA USER STATIONS	(PDUS)
3.0	IMAGE DATA	
3.1	METEUSAT	
3.2	GOES-E	
4.0	111011	
4.0	HIGH RESOLUTION FORMATS	· · · · · · · · · · · · · · · · · · ·
7.1	rik-iomats originating from ESOC	
	1.1 Visible channel data in A- and B-for	mats
	1.2 A-Iormat	
4.1	D-IOTMAL	
7.2	Tik-formats originating from CMS-Lann	on
4.2	2.1 X-format	
5.0	DIGITAL TRANSMISSION FORMAT	s 1
J. 1	Definition of terms	
J. L	r offinal construction	
5.3	Elements of subframes	1
5.3.	1 Synchronisation Word	18
5.3.	2 ID-Word	19
5.3.	3 Label	20
5.3.	4 Identification	
5.3.	5 Interpretation Data	· · · · · · · · · · · · · · · · · · ·
5.3.	6 Fillers	· · · · · · · · · · · · · · · · · · ·
5.3.	7 Grid Information	· · · · · · · · · · · · · · · · · · ·
.4 (Construction of subframes	·········· 22
5.4.	l Heading subframes	23
5	4	23
	4 1.2 Heading subframes of B. and V. Co	23
	2 Data subframes	ormats 23
5	4.7.1 A.E	
	Tormat data subitatnes	24
	- 2 2 A TOTHIAL GATA SUBTRAMES	24
J.7.J	Conclusion subframes	36

5.5	Sets of data subframes	. 25
6.0	TRANSMISSION FORMATS	. 27
7.0	THE INTERPRETATION DATA BLOCK	
7.1	The Format of Integer, Logical and Real Values	. 33
7.2	Calibration Data	33
7	.2.1 Description of permeter in the Calbarra C	34
73	.2.1 Description of parameters in the Calibration Section	35
7.5	Spacecraft Operations Data	37
- /·	.3.1 Description of the parameters in the Spacecraft Operations Section	38
7.4	Imagery Data	41
/.	.4.1 Description of parameters in the Imagery Section	42
7.5	Administrative messages	49
8.0	METEOSAT DISSEMINATION SCHEDULE	
8.1	Key to the METEOSAT Dissemination Schedule	51
8.2	Availability of High Resolution formats	53
	Availability of High Resolution formats	54
9.0	TECHNICAL SPECIFICATION OF A PRIMARY DATA USER STATION (PDUS)	
9.1		
	1 DCS System Recommendations	56

List of Illustrations

Figure	: I.	The METEOSAT Missions.	
Figure	2.	The B-format.	
Figure	3.	The X-format.	
Figure	4.	The area of the A-format (infrared).	1
Figure	5.	The area of the Λ -format (water vapour).	1
Figure	6.	The area of the B-format (visible).	1
Figure	7.	Representation of the frame/subframe relationship.	1.
Figure	8.	Format construction.	11
Figure	9.	Heading subframe of the A-formats.	2
Figure	10.	Heading subframe of B- and X-formats.	2
Figure	11.	The A-format data subframe.	2
Figure	12.	The B- and X-format data subframe.	2
Figure	13.	Structure of the BI, BV, BW formats.	2
Figure	14.	Structure of the BIW, BIVW, BIV formats.	29
Figure	15.	Structure of the AI, AV, AW formats.	3(
Figure	16.	Structure of the AIVII, AIVW. AIW formats.	3
Figure	17.	Structure of the AIV format.	3:
Figure	18.	Inertial co-ordinate system	39
Figure	19.	The position of the Earth as seen during scanning.	44
Figure	20.	Co-ordinate system used to describe image deformation.	4
Figure	21.	Maximum deformation differences (XDDIFM, YDDIFM, XSCM, YSCM).	46
Figure	22.	Image conditions (CONDS).	48
Figure	23.	Example of the METEOSAT Dissemination Schedule	52
Figure	24.	Block diagram of a typical PDUS	< 4

List of Tables

lable	1. Possible A-format combinations.	
Table	2. Possible B-format combinations.	
Table	3. Interpretation data block: calibration section.	- 5
Table	4. Interpretation data block: spacecraft operations section.	34
Table	5. Interpretation data block: imagery data section.	ر د
Table	6. Interpretation data block: administrative message format.	4 !
	Total Communicative Message Ionnat.	49

List of Illustrations

Figure	1.	The METEOSAT Missions.	
Figure	2.	The B-format.	. (
Figure	3.	The X-format.	16
Figure	4.	The area of the A-format (infrared).	1
Figure	5.	The area of the Λ -format (water vapour).	1
Figure	6.	The area of the B-format (visible).	1
Figure	7.	Representation of the frame/subframe relationship.	1
Figure	8.	Format construction.	18
Figure	9 .	Heading subframe of the A-formats.	2:
Figure	10.	Heading subframe of B- and X-formats.	2:
Figure	11.	The A-format data subframe.	24
Figure	12.	The B- and X-format data subframe.	24
Figure	13.	Structure of the BI, BV, BW formats.	28
Figure	14.	Structure of the BIW, BIVW, BIV formats.	29
Figure	15.	Structure of the AI, AV, AW formats.	30
		Structure of the AIVII, AIVW, AIW formats.	
		Structure of the AIV format.	
		Inertial co-ordinate system	
Figure	19.	The position of the Earth as seen during scanning.	
		Co-ordinate system used to describe image deformation.	
Figure	21.	Maximum deformation differences (XDDIFM, YDDIFM, XSCM, YSCM)	46
Figure	22.	Image conditions (CONDS).	48
		Example of the METEOSAT Dissemination Schedule	
Figure	24.	Block diagram of a typical PDUS	56

Preface

The aim of this document is to provide the information required to design and operate a Primary Data User Station (PDUS) and thereby receive and process METEOSAT High Resolution Dissemination formats.

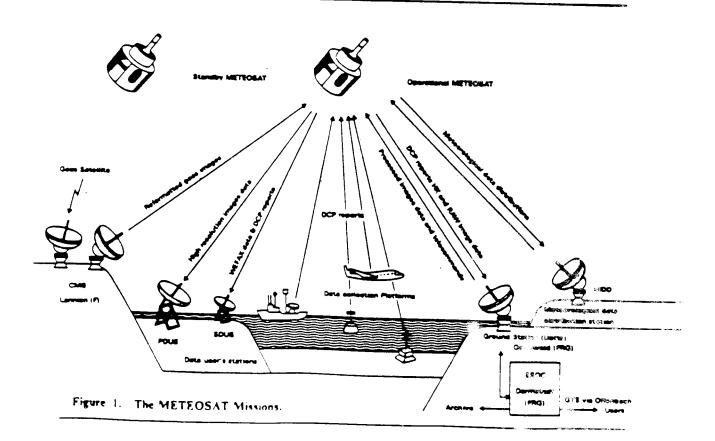
High resolution digital data are designed to be primarily a high quality service for the EUMETSAT Meteorological Services, which fund the METEOSAT System. Readers should note that technical control of these data is currently being considered in order to ensure a proper data access in accordance with EUMETSAT's data policy.

This document has been established by the METEOSAT Exploitation Project in the European Space Operations Centre (ESOC), Darmstadt, Federal Republic of Germany, on behalf of EUMETSAT, the European organisation for the exploitation of meteorological satellites based also in Darmstadt, Federal Republic of Germany. ESOC is part of the European Space Agency (ESA), Paris, France, which operates the METEOSAT Operational Programme on behalf of EUMETSAT.

1.0 INTRODUCTION

METEOSAT dissemination is the process whereby image data and some other meteorological information are relayed via the spacecraft to the user community (see Figure 1).

The spacecraft has two dedicated dissemination channels operating in S-band at 1691.0 MHz and 1694.5 MHz. Two forms of transmission are employed; conventional analogue transmissions (WEFAX), and digital transmissions (HR = High Resolution). The WEFAX transmissions are in the APT (Automatic Picture Transmission) format and are intended for users who wish to use the data in a subjective manner. Information about this type of transmission can be obtained from the document entitled "METEOSAT WEFAX Transmissions". The HR digital data are in a format specific to METEOSAT and are designed for users requiring data with the full radiometric and spatial resolution in a form suitable for further local data processing.



2.0 PRIMARY DATA USER STATIONS (PDUS)

METEOSAT High Resolution (HR) transmissions can be received by Primary Data User Stations (PDUS). Technical specifications and recommendations for the construction of a PDUS are provided in a later section of this document, however, in its simplest form a PDUS will comprise of:

- parabolic antenna;
- low noise amplifier;
- down converter:
- receiver with demodulator:
- bit- and frame-synchronisers;
- image processor.

The images distributed via IIR-dissemination have the full resolution of the original image and are therefore eminently suitable for further quantitative processing. The complexity of the image handling system will be dependent on the level of processing envisaged by the user. Typical processing may include remapping of images to different projections, derivation of meteorological products for local areas and the merging of image data with that from other sources, e.g. radar data, conventional meteorological data, forecasts and analyses. It is also probable that a high resolution hard copy device and/or a colour display would be required.

These suggestions only provide an indication of the type of processing which may be performed on HR-dissemination formats. The cost of a PDUS station, beyond the essential elements of the ground receiving station, will mainly depend on the complexity of the image handling system

3.0 IMAGE DATA

The images disseminated in the form of HR-formats originate from two different sources: METEOSAT and the GOES-E spacecraft

3.1 METEOSAT

METEOSAT takes images in three spectral bands:

```
0.5 - 0.9\mum - visible band;
5.7 - 7.1\mum - infrared water vapour absorption band;
10.5 - 12.5\mum - thermal infrared (window) band.
```

The infrared and water vapour images are composed of 2500 lines, each of 2500 pixels, whilst the visible image consists of 5000 lines of 5000 pixels. The corresponding spatial resolutions at the sub-satellite point are 5 km and 2.5 km, respectively. Scanning of the Earth nominally takes place every half-an-hour, providing images in all three spectral channels. Data from all spectral channels are coded in eight bits corresponding to 256 grey levels (0 - 255).

3.2 GOES-E

The GOES-E spacecraft is nominally located in geostationary orbit at 75 W, however, when only one GOES spacecraft is operational it may be moved to a more westerly location. Images are disseminated from the following spectral bands:

```
0.55 - 0.75\mum visible channel;
10.50 - 12.60\mum infrared channel.
```

The instrument on board GOES is known as the Visible and Infrared Spin Scan Radiometer (VISSR). The imager scans one infrared line for each spin of the satellite, providing approximately 7 km resolution at the subsatellite point. In the visible channel eight lines are scanned simultaneously giving a maximum resolution of about 1 km at the sub-satellite point, however, visible channel data disseminated through METEOSAT has the scan lines combined in groups of two giving an approximately 2 km resolution.

4.0 HIGH RESOLUTION FORMATS

METEOSAT HR-formats originate from the European Space Operations Centre (ESOC) located in Darmstadt, Federal Republic of Germany, while those from GOES-E are received, reformatted and transmitted to METEOSAT for dissemination by the Centre de Météorologie Spatiale in Lannion, France. Formats from both centres are transmitted in the same mode and the transmission speed is 166.66 kbits/s.

4.1 HR-formats originating from ESOC

All raw image data received at the computer processing centre in ESOC are pre-processed e.g. to correct for registration errors caused by the different fields of view of the detectors. In addition, to compensate for the motion of the satellite, geometrical processing is performed. This means that for the user the images always appear to be taken from a satellite at its nominal location, i.e. the received image is located to a fixed reference frame. The end result of geometrical processing is known as a rectified image and has the advantage that, when images are replayed in ar animation sequence, there is no discernible movement of geographical features. Space view data are artificially set to the value 0.

The geometrical image processing is based on an image geometry model which derives its inputs from horizon detection on the infrared image and results in a very accurate fine attitude determination. The rectification of the current image is calculated from the deformation vector field obtained from the geometry model of the current image. This image geometry model is updated dynamically and describes the movement of an image related reference frame as a function of radiometer scan step. This allows the calculation of a deformation vector for every image point on the Earth.

4.1.1 Visible channel data in A- and B-formats

In the A- and B-formats the VIS channel data is indicated as coming from the "southern" (VISs) or "northern" (VISn) detector. This means that during the scanning of the Earth the VISs detector scans the line immediately to the south of the VISn detector, in terms of geographical coordinates. The nominal modes of operation of the visible detectors are VIS1 plus VIS2, with VIS3 plus VIS4 as back-up, however, other combinations are possible, e.g. VIS2 plus VIS4, in the case where one of the detectors in the nominal pairs is defective.

If the required visible channel is not active (e.g. due to a spacecraft failure) the dissemination soft-ware uses the following rules:

- If one visible channel (south or north) is required, and data for only one visible channel are available, then use that data even if it is not that for the required channel.
- If two visible channels are required, and data for only one visible channel is available, then duplicate the data for that channel.

4.1.2 A-format

The A-format represents the whole earth disk as seen by METEOSAT's visible, infrared and water vapour detectors (Figures 4 and 5). The image information of one or more detectors can be transmitted in a given format. All possible combinations of Λ -formats are given in Table 1.

Format	VIS	IR	wv	Transmission time (min)
AV	VISs (2500 lines • 5000 pixels) + VISn (2500 lines • 5000 pixels)		•	23.5
AI	·	2500 lines * 2500 pixels	•	6
AW	·	•	2500 lines * 2500 pixels	6
AIV	as AV	as Al		29.3
AIW	•	as Al	as AW	
ΛΙVW	VISn (2500 lines • 5000 pixels)	as Al	as AW	23.5
AIVH	VISs (2500 lines * 2500 pixels) (Data reduction by taking only each second pixel within the line)	as Al	-	11.8

Table 1. Possible A-format combinations.

With the exception of the AIVH format, all images have the same resolution as the raw image. The A-format consists of 5000 lines of 5000 pixels in the visible channel and 2500 lines of 2500 pixels in the infrared and water vapour channels. In the AIVH formats the visible channel data is reduced to the resolution of the infrared and water vapour channels by only using data from one visible channel and sampling every second pixel in a line

J. Carlo

4.1.3 B-format

The B-format covers the European, North African and Middle East regions and may contain information from one or more of METEOSAT's detectors. Table 2 shows all possible B-formats which could be produced and Figures 2 and 6 indicate the geographical area covered by the B-format.

Format	VIS	IR	wv	Transmission time (min)
BV 	VISa (625 lines * 2500 pixels) + VISn (625 lines * 2500 pixels)	•	•	3.1
BI	·	625 lines * 1250 pixels	•	0.9
BW	·	•	625 lines * 1250 pixels	0.9
BIV	as BV	as Bi	1.	3.8
BIW	•	as BI	as BW	1.6
BIVW	VISn (625 lines * 2500 pixels)	as BI	as BW	3.1

Table 2. Possible B-format combinations.

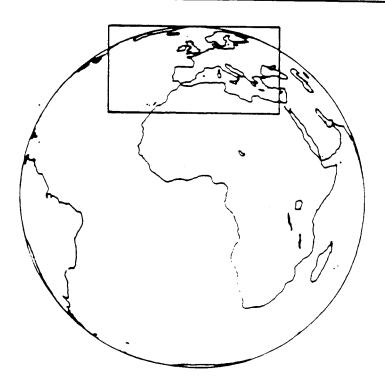


Figure 2. The B-format.

METEOSAT scans the Earth from south to north and from cast to west. The starting line of 8-formats is 1810, the last line is 2434. For the IR and WV formats the first pixel in a line is 626, the last 1875. The first pixel within a visible line is 1252 and the last 3751.

4.2 HR-formats originating from CMS-Lannion

4.2.1 X-format

The X-format covers the North and South American continents and contains infrared, or a combination of visible and infrared, image data from the NOAA GOES-E satellite. The X-format consists of 2500 lines of 2500 pixels in the visible channel and 1250 lines of 1250 pixels in the infrared channel. The resulting resolutions at the sub-satellite point are 2 km for the visible channel data and 7km for the infrared data. The transmission of an XI-format takes place during one dissemination time slot (4 minutes), the XIV-format during two dissemination time slots. The area of an X-format is indicated in Figure 3.

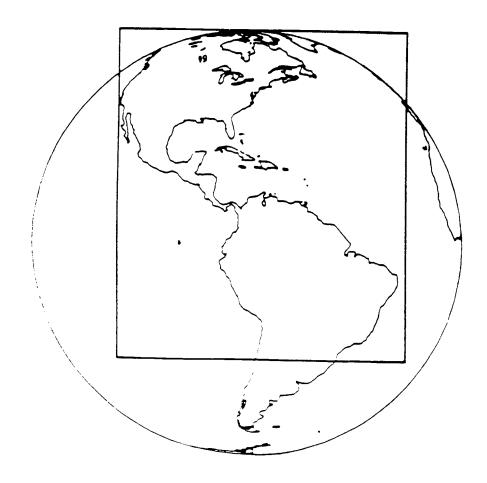


Figure J. The X-format.

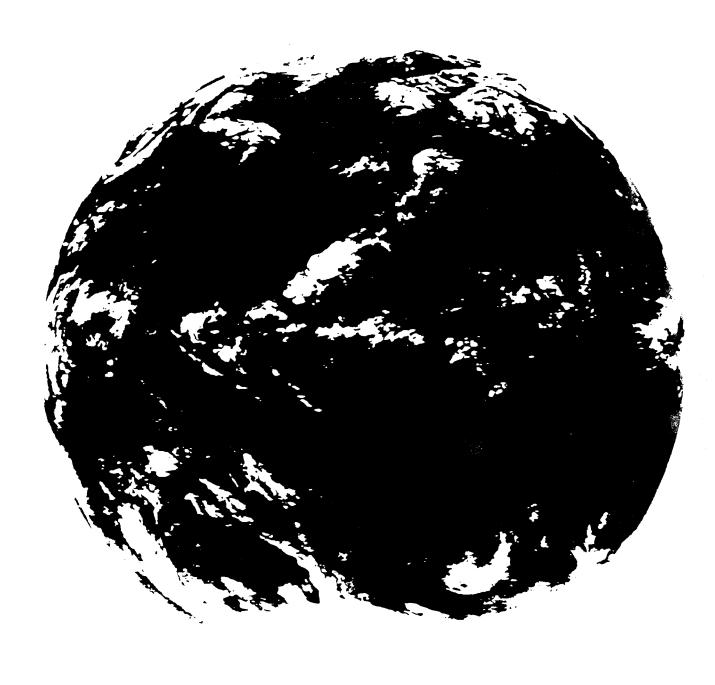


Figure 4. The area of the A-format (infrared).

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Figure 5. The area of the Λ -format (water vapour).

Figure 6. The area of the B-format (visible).

5.0 DIGITAL TRANSMISSION FORMATS

5.1 Definition of terms

Formats are structured into subframes and frames; the following definitions apply:

FORMAT:

Data set used for the transmission of image information, e.g.

A-formats: AV, AI, AIVW
B-formats: BV, BI, BIVW
X-formats: XI, XV, XIV

FRAMES:

Each data frame consists of 364 words of 8 bits.

SUBFRAMES:

An A-format subframe consists of 8 frames whilst B- and X-format subframes consist of 4 frames.

1. 364 words of 8 bits	Frame		
3.		B-format	
4.		subframe	
5.			A-format subframe
0.			300112110
8			

Figure 7. Representation of the frame/subframe relationship.

5.2 Format construction

A complete format consists of the following sets of subframes:

- Heading subframes;
- (2) Image data and grid information subframes;
- 3. Conclusion subframes.

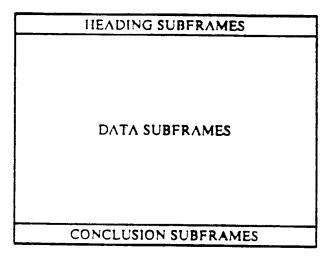


Figure 8. Format construction.

5.3 Elements of subframes

Before explaining the construction of subframes in detail it is first necessary to describe the other data fields within the subframes. These data fields are:

- Synchronisation Word;
- ID-Word;
- Label;
- Identification:
- Interpretation Data.
- Fillers: SP1, SP2, SP4, SP5 and SP6,
- Grid Information.

5.3.1 Synchronisation Word

Each frame starts with a synchronisation word. This word is identical in all frames and all sub-frames. It consists of the following three bytes:

(MSB) 00000101 00001100 11011111 (LSB)

MSB = Most Significant Bit;
LSB = Least Significant Bit.

5.3.2 ID-Word

The ID-word is the fourth byte of each frame. It consists of 8 bits and the value of the ID-word is increased by 1 for each frame within a subframe. The first ID-word of the first frame in each A-format subframe is:

(MSB) 01110000

Since A-format subframes consist of 8 frames the ID-word of the last frame in each Λ -format subframe is:

(MSB) 01110111

The ID-word of the first frame in each B-format and X-format is:

(MSB) 00110000

Since the B- and X-format subframes consist of 4 frames, the ID-word of the last frame in each B- or X-format subframe is:

(MSB) 00110011

By interpreting the ID-word it is possible to distinguish between Λ - and B- or X-formats. The difference between them is in bit 2 of the ID-word which is

1 for A-formats
and
0 for B- and X-formats

5.3.3 Label

The label consists of 24 bytes and is present in the first frame of each subframe. The contents of the label are as follows:

Byte Description

- 1 + 2 Number of frames in the subframe: 8 for A-formats, 4 for B- and X-formats.
- Total number of subframes in the format counting the 42, 84 or 90 header subframes as 1 and all conclusion subframes as 1. (The total number of subframes is therefore 2 plus the number of data subframes).
- 5 + 6 Current subframe number. For all header and conclusion subframes the current subframe number is 1. The counting of data subframes starts with 0 for the first data subframe.
- 7 + 8 Image line number. This is zero in the header and conclusion subframes. In Λ- and X-formats the first image line number is 1 and in B-formats is 1810. In all cases the image line number increases by 1 for each set of subframes that corresponds to a single line.
- 9 12 Image number from the start of the mission.
- \star 13 Format indicator (hexadecimal); $\Lambda = 00$, B = FF, X = 0F.
 - 14 to 17 Indicators of spectral information in the subframe.

Byte 14 = VISs.

Byte 15 = VISn.

Byte 16 = IR.

Byte 17 = WV.

These indicators are set to the following hexadecimal values:

00 if corresponding data are not represented in the subframe FF if corresponding data are represented in the subframe.

Since subframes containing full resolution VIS information contain only half lines, bytes 14 and/or 15 are used in data subframes as follows

FO if 1st half of VIS line is present in the data subframe. OF if 2nd half of VIS line is present in the data subframe.

set to FF.

18 Indicator of grid information:

```
00 if no grid information is present.

OF if reference grid information is present.
```

- Spare bytes, set to 00.
- 20 Indication of scan direction.

```
00 if scan is S-N and E-W. F0 if scan is N-S and E-W. OF if scan is S-N and W-E. FF if scan is N-S and W-E.
```

Normally byte 20 has the hexadecimal value 00

21 - 24 Spare bytes, set to 00.

5.3.4 Identification

The identification field is present in the first frame of all heading and conclusion subframes.

Byte Description

1 + 2 Satellite indicator. The origin of the image data contained in the transmitted format is given by the satellite indicator.

```
GOES = 0000 (hexadecimal)

METEOSAT 1 = D4E3 = MT (EBCDIC CODE)

METEOSAT 2 = D4F2 = M2 ( ... ...)

METEOSAT n = D4Fn = Mn ( ... ...)
```

- 3 + 4 Year of image acquisition (e.g. 1989) coded as a binary value.
- 5 + 6 Day of year (e.g. 346) of image aquisition coded as a binary value.

- 7 + 8 Nominal image time. This information is coded in binary (BCD Binary Coded Decimal) and indicates hour and minute of completion of image acquisition e.g. 0100 (slot 02), 1430 (slot 29).
- 9 20 As for the label in header subframes
- 21 32 Spare bytes, set to 00.

5.3.5 Interpretation Data

The Interpretation Data block is present in the first four frames of each header and conclusion subframe. It consists of 1360 bytes and provides information on calibration, satellite orbit, geographical correction parameters and administrative messages. A full description of the Interpretation Data block is given in Section 7

5.3.6 Fillers

Fillers are included in the data transmission in order to keep subframes to a fixed structure. They are as follows:

SP1 = 8 bytes, all set to 00.

SP2 = 16 bytes, all set to 00.

SP4 = 360 spare bytes, all set to 00.

SP5 = 40 bytes, all set to 00.

SP6 = 8 bytes, all set to 00.

5.3.7 Grid Information

Grid information is enclosed in each data subframe. The n^{th} bit of the grid information corresponds to the n^{th} byte of the image data of the same data subframe. The bit is set to 1 if grid point data is available.

The grid information of A-formats consists of 316 bytes (2528 bits). The first 2500 bits are the grid information corresponding to 2500 image pixels within the subframe. 28 spare bits are set to 0

The grid information of B- or X-formats consists of 158 bytes (1264 bits). The first 1250 bits represent the grid information corresponding to the 1250 image pixels transmitted within the subframe, 14 spare bits are set to 0.

See promision

5.4 Construction of subframes

5.4.1 Heading subframes

F. MES

5.4.1.1 Heading subframes of A-formats

The heading subframes of A-formats consist of 8 frames of 364 bytes each. The structure is as follows.

----- 364 bytes

SYNC ID LABEL SPI IDENTIFICATION SP₂ INTERPRETATION 24 16 DATA (280 bytes) SYNC ID INTERPRETATION DATA (360 bytes) SYNC ID INTERPRETATION DATA (360 bytes) SYNC ID INTERPRETATION DATA (360 bytes) SYNC ID SP4 (BLANK FILL) SYNC ID SP4 SYNC ID SP4 SYNC ID SP4

Figure 9. Heading subframe of the Λ -formats.

The heading subframe is transmitted 42 times for each A-format.

5.4.1.2 Heading subframes of B- and X-formats

The heading subframes of B- and X-formats consist of 4 frames each of 364 bytes. The structure is as follows:

----- 364 bytes -

SYNC ID LABEL SPI IDENTIFICATION SP2 INTERPRETATION DATA (280 bytes) SYNC ID INTERPRETATION DATA (360 bytes) SYNC ID INTERPRETATION DATA (360 bytes) SYNC INTERPRETATION DATA (360 bytes)

Figure 10. Heading subframe of B- and X-formats.

The heading subframe is transmitted 84 times for B-formats and 90 times for X-formats.

5.4.2 Data subframes

5.4.2.1 A-format data subframes

A-format data subframes consist of 8 frames of 364 bytes each. The structure is as follows:

364 bytes -----

SYNC	ID	LABEL SP5 D	111 (296 bytes)	
SYNC	ID	DATA (360 bytes)		
SYNC	ID	DATA (360 bytes)		
SYNC	ID	DATA (360 bytes)		
SYNC	ID	DATA (360 bytes)		
SYNC	ID	DATA (360 bytes)		
SYNC	ID	DATA (360 bytes)		
SYNC	ID	DATA (44 bytes)	(316 bytes) GRID	

Figure 11. The A-format data subframe.

Each A-format data subframe includes 2500 bytes of image information. This corresponds to 2500 image pixels and represents either one complete line of an IR or WV image or half a line of the image from one VIS channel (in the AIVII format the VIS channel data is from one line but only every other pixel).

5.4.2.2 B- and X-format data subframes

B- and X-format data subframes consist of 4 frames of 364 bytes each.

The structure is as follows:

----- 364 bytes -----

SYNC	ID	LABEL SP6 DATA (328 bytes)
SYNC	ID	DATA (360 bytes)
SYNC	ID	DATA (360 bytes)
SYNC	ID	DATA (202 bytes) (158 bytes) GRID

Figure 12. The B- and X-format data subframe.

Each B-format data subframe includes 1250 bytes of image information. This corresponds to 1250 image pixels and represents one complete B-format line of an IR or WV image or half a B-format line of an image from one VIS channel.

5.4.3 Conclusion subtrames

The conclusion subframe is a copy of the heading subframe. The number of conclusion subframes transmitted is:

```
for all A-formats
for BI. BIV, BW, BVW and X-formats;
```

2 for P:w. BIVW and BV.

5.5 Sets of data subframes

 Λ set of data subframes represents one image line of all spectral channels included in the individual transmission format. Λ set of data subframes therefore consists of one or a combination of the following elements:

- one subframe of IR data;
- two subframes of VISs data:
- two subframes of VISn data;
- one subframe of WV data.
- Consequently, the number of data subframes in a set of subframes varies in accordance with the number of spectral channels for which information is included in the transmission format. The formats include the following information:

AI, BI, XI	subframe	IR only
AIV, BIV	subframes	IR + VISs + VISn
AIW, BIW	subframes	IR + WV
AIVW, BIVW	subframes	IR + WV + VISs
AW, BW	subframe	WV only
AV, BV	subframes	VIS only (VISs + VISn)
AIVH	subframes	Reduced VISs + IR
XIV	subframes	IR + VIS

Within the sets of subframes the data subframes from the different spectral channels are transmitted in the sequence

IR, VISs, VISn, WV.

6.0 TRANSMISSION FORMATS

A list of available transmission formats is given in Sections 4.1.1 (A-formats), 4.1.2 (B-formats) and 4.2.1 (X-formats). Examples of the construction of some transmission formats are shown in the following figures. In the case where information from one visible channel is not available, the information from the other visible channel is repeated in formats AV, BV, AIV and BIV.

BI-format

Heading sub- frames	IR line 1810	1R line 1811	IR line 1812	IR line 1813	1R line 1814	1R line 1815	IR line 1816	IR line 1817	IR line 1818		IR line 2429	IR line 2430	IR line 2431	IR line 2432	IR line 2433	IR line 2434	Conclusion
336F						625	subí	ramo	:5 =	2500 fra	mes				·—-		1SF = 4F

BV-format

Heading sub- frames	VISs 1/2 line 1810	VISs 1/2 line 1810	VISn 1/2 line 1810	VISn 1/2 line 1810	VISs 1/2 line 1811	VISs 1/2 line 1811	VISa 1/2 line 1811	VISn 1/2 line 1811	VISs 1/2 line 1812		VISn 1/2 line 2433	VISa 1/2 line 2433	VISs 1/2 line 2434	VISs 1/2 line 2434	VISn 1/2 line 2434	VISn 1/2 line 2434	Conclusion
84SF 336F					4	2500	subi	rame	:5 =	10000 fr	ame:	3		<u> </u>	<u>[</u>	l	2SF = 8F

BW-format

Heading sub- frames	WV line 1810	WV line 1811	WV line 1812	WV line 1813	WV line 1814	WV line 1815	WV line 1816	WV line 1817	WV line 1818	WV line 3429		WV line 2431	WV line 2432	WV line 2433	WV line 2434	Conclusion
84SF 336F		•		<u> </u>	<u> </u>	625	subf	rame	:s =	2500 frame	3	<u>!</u>				ISF = 4F

Figure 13. Structure of the BI, BV, BW formats.

Dive iuilliai

Heading sub- frames	IR line 1810	WV line 1810	IR line 1811	WV line 1811	IR line 1812	WV line 1812	IR line 1813	WV line 1813	IR line 1814		IR line 2432	WV line 2432	IR line 2433	WV linc 2433	IR line 2434	WV line 2434	Conclusion
84SF = 336F						1250	sub	fram	es =	5000 fra	ımes						2S F = 8 F

BIVW-format

Heading sub- frames	IR line 1810	VISa 1/2 line 1810	VISa 1/2 line 1810	WV line 1810	IR line 1811	VISa 1/2 line 1811	VISn 1/2 line 1811	WV line 1811	IR line 1812		VISa 1/2 line 2433	WV line 2433	IR line 2434	VISn 1/2 line 2434	VISn 1/2 line 2434	WV line 2434	ேவctusion
84SF 336F						2500	subí	rame	:5 =	10000 fr	ame:	3	L,	<u> </u>	<u>L</u>		2SF 8F

BIV-format

Heading sub- frames	IR line 1810	VISs 1/2 line 1810	VISs 1/2 line 1810	VISn 1/2 linc 1810	VISn 1/2 line 1810	IR line 1811	VISs 1/2 line 1811	VISs 1/2 line 1811	VISa 1/2 line 1811		VISn 1/2 line 2433	IR line 2434	VISs 1/2 line 2434	VISs 1/2 line 2434	VISa 1/2 line 2434	VISn 1/2 line 2434	(Onclusion
84SF = 336F						3125	subí	iramo	:5 =	12500 fr	ame:	3		<u> </u>	<u> </u>	<u></u>	2SF = 8F

Figure 14. Structure of the BIW, BIVW, BIV formats

AI-format

Heading sub- frames	IR line 1	IR line 2	IR line 3	IR line 4	IR line 5	IR line 6	IR line 7	IR line 8	IR line 9		IR line 2495	IR line 24%	IR line 2497	IR line 2498	IR line 2499	IR line 2500	Conclusion
42SF = 336F						2500	subí	Lame	:5 =	20000 fr	ame:	3			<u> </u>		ISF 8F

AV-format

Heading sub- frames	VISs 1/2 line 1	VISs 1/2 line 1	VISn 1/2 line 1	VISa 1/2 line 1	VISs 1/2 line 2	VISs 1/2 line 2	VISn 1/2 line 2	VISn 1/2 line 2	VISs 1/2 line 3		VISn 1/2 line 2499	VISn 1/2 line 2499	VISs 1/2 line 2500	VISs 1/2 line 2500	VISn 1/2 line 2500	VISn 1/2 line 2500	Conclusion
42SF = 336F					1	0000	sub	(ram	es =	80000 fr	an.	:5					ISF = 8F

AW-format

Heading sub- frames	WV line 1	WV line 2	WV line 3	WV line 4	WV line 5	WV line 6	WV line 7	WV line 8	WV line 9		WV line 2495	WV line 2496	WV line 2497	WV line 2498	WV line 2499	WV line 2500	Conclusion
42SF = 336F		-	_			2500	subf	[rame	:5 =	20000 fra	mes	ı			-		ISF
																	8F

Figure 15. Structure of the AL AV, AW formats.

AIVH-format

Heading sub- frames	IR line 1	VISsf1 line 1	IR line 2	VISsH line 2	IR line 3	VISsH line 3	IR line 4	VISsII line 4	IR line 5		IR line 2498	VISsII line 2498	IR line 2499	VISsII line 2499	IR line 2500	VISsIT tine 2500	Conclusion
42SF = 336F						5000	subf	rame	:5 =	40000 fran	nes						ISF =

AIVW-format

Heading sub- frames	IR line 1	VISn 1/2 line 1	VISn 1/2 line 1	WV line 1	IR line 2	VISn 1/2 line 2	VISn 1/2 line 2	WV line 2	IR line 3		VISn 1/2 line 2499	WV line 2499	IR line 2500	VISn 1/2 line 2500	VISn 1/2 line 2500	WV line 2500	Conclusion
42SF = 336F		•		<u></u>	ì	0000	sub	lram	es =	80000 f	ramo						ISF
																	8F

AIW-format

Heading sub- frames	IR fine 1	WV line 1	IR line 2	WV line 2	IR line 3	WV line 3	IR line 4	WV line 4	IR line 5		IR line 2498	WV line 2498	IR linc 2499	WV line 2499	IR line 2500	WV line 2500)	Conclusion
42SF 336F						5000	subí	rame	:5 =	40000 fr	ames	l e	<u> </u>	<u> </u>	<u> </u>		ISF

Figure 16. Structure of the AIVII, AIVW, AIW formats.

AIV-format

Heading sub- frames	IR line 1	VISs 1/2 line 1	VISs 1/2 line 1	VISa 1/2 line 1	VISn 1/2 line 1	IR line 2	VISA L 2 line 2	VISs 1,2 line 2	VISn 1/2 line 2	VISn 1/2 line 2499	IR line 2500	VISs 1/2 line 2500	VISs 1/2 line 2500	VISn 1/2 line 2500	VISn 1/2 line 2500	Conclusion
42SF 336F					13	2500	suhí	rame	:5 =	100000 fram	25	1	<u></u>	L		ISF = 8F

Figure 17. Structure of the AIV format.

7.0 THE INTERPRETATION DATA BLOCK

The Interpretation Data block contains 1360 bytes and is organised in the following way:

- 1. Calibration Data (104 bytes);
- 2. Spacecraft Operations Data (128 bytes);
- 3. Imagery Data (328 bytes);
- 4. Administrative Message (800 bytes).

7.1 The Format of Integer, Logical and Real Values

Integer values (I*2, I*4)

Two-byte and four-byte integer values are represented as 16-bit and 32-bit binary numbers respectively. The first bit indicates the sign. A positive integer is represented by a 0 in the sign bit and a binary true number representation in the remaining bits. A two's complement with 1 in the sign bit represents a negative integer. All bits contain 0's to represent zero.

Logical values (L*1)

One-byte logical values are represented as 8-bit binary numbers. A true value has 1 in the least significant bit bit with all remaining bits set to 0. A false value has all bits set to 0.

Real values (R*4)

Real values are represented as 32-bit floating point numbers. The first bit indicates the sign. The seven bits following the sign bit indicate the exponent part. The mantissa consists of 24 bits following the exponent part, each four bits representing a hexadecimal digit. The hexadecimal radix point of the mantissa is to the left of the most significant mantissa digit. The exponent part is a number in the range 0 to 127. It represents an exponent in the range -64 to 63, after subtraction of 64 from the actual exponent

0			24	31
S 1	E	М	٠	•

Double precision values (R*8)

Double precision values are represented as a 64-bit floating point number. Its sign and exponent part are similar to those described above. The mantissa consists of 56 bits following the exponent part and has 14 hexadecimal digits, each four bits representing a digit.

	8					
S E	:	*****	 H		••••••	 I
	• • • • • • • • •		 	••••••		 •

- S: sign (0 = positive, 1 = negative).
- E: exponent (value obtained by adding 26 to the actual exponent).
- M: hexadecimal mantissa part with the radix point to the left of bit 8.

Value = Man. 16exp.

7.2 Calibration Data

Offset	Name	Туре	Length	Content
0	BBCIIR	ASCII	6	BB IR count for space view calibration XXX.XXX (XXXXXX is sent)
6	BBSDII	ASCII	3	Standard deviation of BB IR count for space view calibration X.XX (XXX is sent)
9	BBCIWV	ASCII	6	BB WV count for space view calibration XXX.XXX (XXXXXX is sent)
15	BBSDIW	ASCII	3	Standard deviation of BB WV count for space view calibration X.XX (XXX is sent)
18	BBIT	ASCII	5	Temperature of 'cold' BB in K (only valid for nominal calibration mode) XXX.XX (XXXXX is sent)
23	BB2T	ASCII	5	Temperature of 'hot' BB in K in either space or nominal calibration mode XXX.XX (XXXXX is sent)
28	BBC2IR	ASCII	6	BB IR count for nominal calibration mode XXX XXX (XXXXXX is sent)

Offset	Name	Туре	Length	Content
34	BBSD21	ASCII	3	Standard deviation of BB IR count for nominal calibration mode X.XX (XXX is sent)
37	BBC2WV	ASCII	6	BB WV count for nominal calibration mode XXX.XXX (XXXXXX is sent)
43	BBSD2W	ASCII	3	Standard deviation of BB WV count for nominal calibration mode X.XX (XXX is sent)
46	TIME	ASCII	ς	Timestamp for BB calibration DDDSS (DDD = day, SS = slot)
51	CALIR	ASCII	ζ	MIFC absolute IR calibration value 0.XXXXX XXXX is sent)
56	IRSPC	ΛSCII	;	IR space count XX.X (XXX is sent)
59	TIME2	ASCII	٢	Limestamp for IR calibration DDDSS (DDD = day, SS = slot)
64	CALWV	ASCII	(MIEC absolute WV calibration value 0.XXXXX (XXXXX is sent)
69	WVSPC	ASCII	1	WV space count XX.X (XXX is sent)
72	TIME3	ASCII	5	Firmestamp for WV calibration DDDSS (DDD = day, SS = slot)
77			1.5	Spares
92	GAINS	ASCII	9	Gains for IR, WV, VIS1 or VIS4, VIS2 or VIS3, GiGiGwGwGvGvG2G2 (Gains are in the range 00-15)
100			4	Spares

Table 3. Interpretation data block: calibration section.

7.2.1 Description of parameters in the Calibration Section

There exist two different methods of performing a black body calibration on board the MOP series of satellites. They are termed 'nominal' calibration and 'space' calibration respectively.

Nominal calibration mode

Two black bodies exist on board the spacecraft, the first is known as the 'cold' black body and will assume the ambient temperature of that part of the radiometer in which it is located. The second black body, known as the 'hot' black body, will be heated to a temperature of approximately 50°C above that of the 'cold' black body. In the nominal mode of operation the 'cold' black body is viewed in order to provide a reference level (DC restore level) above which the response of the 'hot' black body is measured. This effectively means that the count telemetred to the ground represents the difference in count between the 'hot' and cold' black bodies. The temperature of the two black bodies is expected to be in the range from 5°C to 35°C for the cold' black body and from 55°C to 85°C for the hot' black body.

Space calibration mode

In this mode use is only made of one black body, the 'hot-black body. However, two options exist, the 'hot-black body can either be heated or can remain at the ambient temperature of the satellite

In the space calibration mode the reference level is provided by the radiometer looking at the space view in both options.

Either method can be used to perform a simultaneous black body calibration of the infrared and water vapour channels. In the situation where heating is applied to the 'hot' black body the analogue signal from the detectors is divided by two for the IR calibration and by eleven for the WV calibration, before being converted into a count!

BBC1IR (BBC1WV)

The raw black body count for the IR (WV) channel observed during a space mode calibration.

BBSDII (BBSDIW)

Standard deviation of the black body count for the IR (WV) channel during a space mode calibration.

BBIT (BB2T)

Temperature of the cold (hot) black body in K. The cold black body temperature is only valid for the nominal calibration mode.

BBC2IR (BBC2WV)

The raw black body count for the IR (WV) channel observed during a nominal mode calibration.

BBSD2I (BBSD2W)

Standard deviation of the black body count for the IR (WV) channel during a nominal mode calibration

CALIR (CALWV)

The absolute MIEC calibration coefficient for the IR (WV) channel which include all adjustments to account for variation of radiometer response. The calibration equation, as used in the MIEC, is for both channels:

Radiance = (count - SPC) * CAL

where SPC and CAL are the space count and calibration coefficient for the corresponding spectral channel

IRSPC (WVSPC)

Average space count for the IR (WV) spectral channels as used within the MIEC.

TIME

This parameter gives the day of the year and slot number at which the black body calibration was performed.

TIME2

This parameter gives the day of the year and slot number at which the IR calibration coefficient became valid.

TIME3

This parameter gives the day of the year and slot number at which the WV calibration coefficient became valid.

GAINS

On board gain settings for the different spectral channels in the order IR, WV, VIS1 (VIS4), VIS2 (VIS3). The gain settings are in the range 00 - 15.

7.3 Spacecraft Operations Data

Offset	Name	Туре	Length	Content
104	DEGSRA	R*8	8	Right ascension attitude south in degrees
112	DEGSDE	R • 8	8	Declination attitude south in degrees
120	DEGNRA	R *8	8	Right ascension north in degrees
128	DEGNDE	R •8	9	Declination attitude north in degrees
136	FINATT	R •8	24	X. Y. Z of refined attitude
160	FARADE	R*8	16	Right ascension and declination of mean refined at-
176	NRSLOT	1•4	.1	Number of vious for refined attitude
180	SPNDUR	R • 4	4	Spin duration minus nominal spin duration (μ,s)
184	FLECL	L·1	1	Eclipse operation

Offset	Name	Type	Length	Content
185	FLDEC	L•1	1	Decontamination
186	FLMAN	L•1	1	Manoeuvre
187	FLMODE	L•I	1	Earth/Sun mode
188	FLIRI	L·I	1	IR1 on
189	FLIR2	L·I	1	IR2 on
190	FLWVI	L•1	1	WVi on
191	FLWV2	L•1	1	WV2 on
192	FLVISI	L•1	1	VIS1 on
193	FLVIS2	L°1	1	VIS2 on
194	FLVIS3	L+I	1	VIS3 on
195	FLVIS4	L•I	1	VIS4 on
196			36	Spares for future use

Table 4. Interpretation data block: spacecraft operations section.

7.3.1 Description of the parameters in the Spacecraft Operations Section

This section of the Interpretation Data provides information on the satellite orbit. The co-ordinate frame used for this data is the geocentric mean equatorial system of date 1950.0 (related to the spring equinox in the year 1950) and often referred to as the y50 co-ordinate frame. The co-ordinate frame is an inertial frame with its origin at the centre of the Earth. The XY plane corresponds to the Earth's equatorial plane and the X-axis points in the direction of the mean vernal equinox (intersection of the Earth's equatorial plane with the ecliptic plane). The Z-axis points in the direction of celestial north (positive towards the star Polaris) and the Y-axis forms the third member of the right-handed XYZ co-ordinate frame. Figure 18 shows the co-ordinate frame y50.

DEGSRA

This value describes the angle between the X-axis and the vector formed by the projection of the attitude vector (spacecraft spin axis) onto the XY-plane. This value is measured during the scanning of the southern horizon of the image.

DEGSDE

This value describes the angle between the attitude vector and its projection on the XY-plane. This angle is measured during the scanning of the southern horizon of the image.

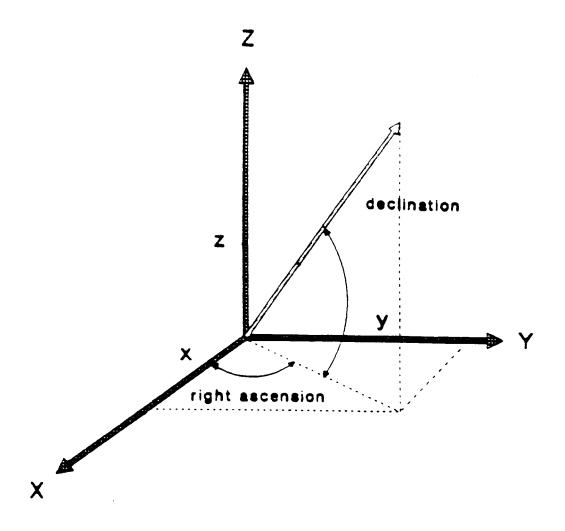


Figure 18. Inertial co-ordinate system y 50: This co-ordinate system is used to describe the satellite position in space and is related to the Spring equinox in 1950.

DEGNRA

This is the value corresponding to DEGSRA but measured during the scanning of the northern horizon of the image.

DEGNDE

This is the value corresponding to DEGSDI has measured during the scanning of the northern horizon of the image

FINATT

These are the X, Y and Z components of the attitude vector (see Figure 18) based on the refined attitude. This is a mean of the attitude vectors over the number of slots given by NRSLOT

FARADE

This is a mean of the values of the right ascension and declination over the number of slots given by NRSLOT.

NRSLOT

This is the number of slots (nominally 48) used to calculate the refined attitude i.e. the number of slots used for calculating the values of FINATT and FARADE.

SPNDUR

This time is measured during the scanning of line 1250, nominally middle of the image, and corresponding to nominal radiometer position 1350. The value transmitted is the measured spin duration in μx minus the nominal spin duration (6.10° μx). If the satellite rotation is faster than nominal this value will be negative.

FL***

The various operation modes are indicated by a L*1 value. If the value is set to 1 the operation mode is active during the acquisition of the image. The flags are:

FLECL - Eclipse operation.

FLDEC - Decontamination of the radiometer

FLMAN - Spacecraft manocuvre.

FLMODE - Satellite is in Earth mode if value is 1, in Sun mode if value is 0.

FLIRI-IRI on.

FLIR2 - IR2 on

FLWVI - WVI on

FLWV2 - WV2 on.

FLVISI - VISI on

FLVIS2 - VIS2 on

FLVIS3 - VIS3 on.

FLVIS4 - VIS4 on.

7.4 Imagery Data

Offset	Name	Туре	Length	Content
232	IMSTAT	L*I	16	Image status I Horizon analysis performed 2 Spin speed fit 3 Orbit offset vector fit 4 Pixel sampling rate fit 5 Attitude refinement iteration based on horizon results 6 Spare 7 Spare 8 Calculation of deformation vector field 9 Completion of geometrical processing 10 Completion of rectification 11 Completion of amplitude processing 12-16 Spare
248	LIMHOR	I*2	24	Southern line, Ist/last pixel Northern line, Ist/last pixel East pixel, south/north line West pixel, south/north line
272	SATDIS	R*8	8	Distance, satellite to earth centre
280	SORBOF	R *8	24	X. Y. Z components of orbit offset vector in nominal image frame when scanning southern honzon
304	NORBOF	R*8	24	X, Y, Z components of orbit offset vector in nominal image frame when scanning northern horizon
328	XDDIFM	R*4	4	Maximum deformation difference X-component within column
332	YDDIFM	R*4	4	Maximum deformation difference Y-component within column
336	XSCM	R • 4	4	Maximum deformation difference X-component within lines
340	YSCM	R • 4	4	Maximum deformation difference Y-component within lines
344	CONDS	L•1	4	Image conditions I Earth within field of view in east west 2 Earth within field of view in south north 3 East west honzon limits within predicted margin 4 South north honzon limits within predicted margin

Offset	Name	Type	Length	Content	
348	LOWDYN	1*2	8	Lowest non-zero count in histogram (-1 is off) (IR 1/2, VIS1/4, VIS2/3, WV1/2)	
356	HIGDYN	1*2	A	Highest non-zero count in histogram (-1 is off) (IR I 2, VIS1/4, VIS2/3, WV1/2)	
364	MVIS1	R *4	1	Mean value of VISI/VIS4	_
368	MVIS2	R*4	1	Mean value of VIS2/VIS3	
372	SNNOM	R • 4	14	Signal to noise ratio in space corners (IR I 2, VIS1/4, VIS2/3, WV1/2)	
388	SNNLIN	1•4	4	Number of lines used to calculate SNNOM	
392	SNREP	R*4	16	ratios eastern part (IR I 2. VIS1/4, VIS2/3, WV1/2)	
408	SNRWP	R*4	16	ratios western part (IR I 2, VIS1/4, VIS2/3, WV1/2)	
424	SWMNEP	R*4	16	Mean noise count eastern part (IR1/2, VIS1/4, VIS2/3, WV1/2)	
440	SWMNWP	R*4	16	Mean noise count western part (IR 1/2, VIS1/4, VIS2/3, WV1/2)	
456	SNMXEP	1*2	8	Maximum space count eastern part (IR1/2, VIS1/4, VIS2/3, WV1/2)	_
464	SNMXWP	1*2	8	Maximum space count western part (IR 1/2, VIS1/4, VIS2/3, WV1/2)	_
472			88	Spare for future use	

Table 5. Interpretation data block: imagery data section.

7.4.1 Description of parameters in the Imagery Section

These data are produced during image processing. The user is provided with information on the level of image processing performed, location of the image frame, deformation parameters and a summary of the results from image amplitude analysis processing.

IMSTAT

These data are represented by 16 L*1 values which are set to 1 if the corresponding processing was successfully completed. The sixteen bytes are dedicated as follows:

- I Honzon analysis performed
- 2 Spin speed fit
- 3 Orbi, offset vector fit.
- 4 Pixel sampling rate fit.

- 5 Attitude refinement iteration based on horizon results
- 6 Automatic landmark registration.
- 7 Actual image frame movement fit based on landmark results.
- 8 Calculation of deformation vector field.
- 9 Completion of geometrical processing.
- 10 Completion of rectification.
- 11 Completion of amplitude processing.
- 12 16 Spare.

LIMHOR

The group of twelve values in LIMHOR indicate the position of the Earth as seen during the scanning of the image. The positions to which the individual parameters refer are shown in Figure 19.

- 1 Southern line: The number of the first scan line containing Earth information.
- 2 First pixel: The column of the first detected pixel in the first line containing Earth information.
- 3 Last pixel: As above, but the column of the last detected Earth pixel in the line.
- 4 Northern line: The number of the last scan line containing Earth information.
- 5 First pixel: The column of the first detected pixel in the last line containing Earth information.
- 6 Last pixel: As above, but the column of the last detected Earth pixel in the line.
- 7 East pixel: The right most pixel (column number) in the image.
- 8 South line: Number of the first scan line containg the right most pixel.
- 9 North line: Number of the last scan line containing the right most pixel.
- 10 West pixel: The left most pixel column number) in the image.
- II South line: Number of the first scan line containing the left most pixel.
- 12 North line: Number of the last scan line containing the left most pixel.

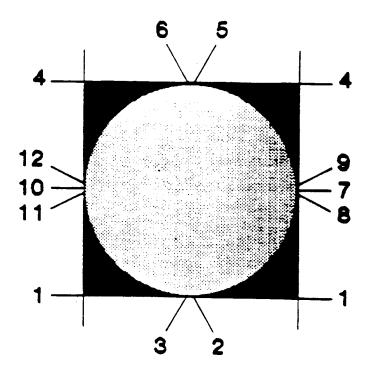


Figure 19. The position of the Earth as seen during scanning.

SATDIS

This double precision value indicates the distance of the satellite to the Earth centre in kilometres at the time of scanning the Earth's centre.

SORBOF, NORBOF

These values indicate the level of changes to the image during image geometry corrections. For the description of the image geometry the following set of co-ordinate systems is used:

- The nominal image frame (NIF) with origin at the nominal satellite position N, Xn-axis pointing towards the Earth centre. Zn-axis pointing northwards, parallel with the Earth's axis of rotation with Zn = 0 being the equatorial plane.
- The actual image frame (AIF) with origin at the momentary satellite position S. Xa-axis pointing towards the Earth. Za-axis coinciding with the attitude vector 'a' and the Earth's centre always being in the plane Ya = 0. Generally this frame is moving with each scan step of the radiometer.

Note: The subscripts a and in refer to AIF and NIF, respectively.

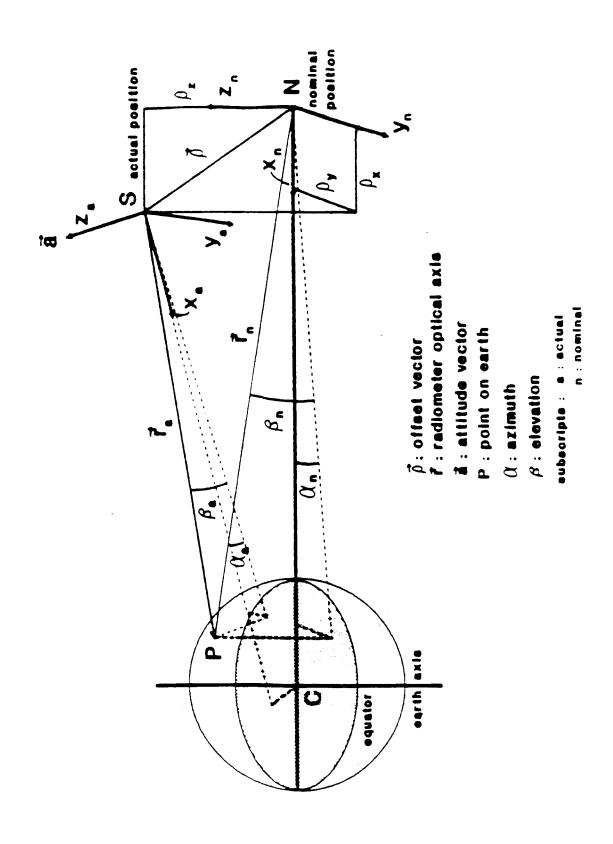


Figure 20. Co-ordinate system used to describe image deformation.: The Earth centre C is always in the planes $Y_0 = 0$ and $Y_0 = 0$

The principle idea of the image geometry model is to adjust the actual image frame at the position of the southern (northern) horizon limit scan and to express its movement (offset vector and Cardan angles) with respect to the nominal image frame as functions of the radiometer scan step. Conservation of angular momentum of the spacecraft (except during manoeuvres) is used for determining an approximate mean attitude vector as starting point for an iterative attitude refinement procedure per image.

Since infrared radiation differs significantly at all times from thermal noise coming from space, even at the polar caps of the Earth, the southern and northern horizon scan limits can easily be extracted from the infrared image. In order to get these positions accurate to a part of a pixel a parabola arc is fitted to the horizons. The radiation threshold value defining the horizon and apparent radius of the Earth are adjustable parameters in order to compensate for the thickness of the atmosphere. Figure 20 illustrates the co-ordinate systems for the description of image deformations. The values ρx , ρy and ρz correspond to the X, Y, Z values in SORBOF, NORBOF.

XDDIFM, YDDIFM, XSCM, YSCM

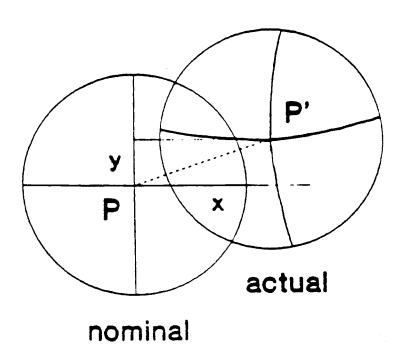


Figure 21. Maximum deformation differences (XDDIFM, YDDIFM, XSCM, YSCM).

Information on the deformation of the image is produced on a grid of 26 lines by 26 columns and the actual image is compared with the nominal reference image (see Figure 21). The maximum differences measured on the 26 x 26 grid are given in the above parameters

CONDS

The first two logical values are set TRUE, when the Earth was in the field of view during scanning. The first logical byte is set TRUE, for E.W scanning, the second is set TRUE, for N.S scanning. Bytes 3 and 4 are set TRUE, when the scanned Earth image was within + - 3 pixels from the predicted values (see Figure 22).

LOWDYN (HIGDYN)

LOWDYN (HIGDYN) indicates the lowest (highest) non-zero count in the histogram for IR1 or IR2, VIS1 or VIS4, VIS2 or VIS3, WV1 or WV2, in that order. If the channel is not active the value is set to -1.

MVISI (MVIS2)

Represents the mean value of image pixels, including space, for the VIS1/VIS4 (VIS2/VIS3) spectral channels.

The four parameters given above provide information which could be used for image enhancement in order to make use of the full dynamic range of hard copy recorders or display units.

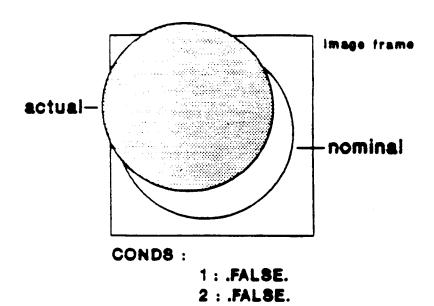
SNNOM, SNNLIN, SNREP, SNRWP, SWMNEP, SWMNWP, SNMXEP, SNMXWP

These parameters provide information on the quality of the radiometric performance during scanning. For each channel the pixel values in space at the four corners of the image are analysed to determine the noise level.

The signal to noise ratio is calculated according to the following formula:

$$S/N = \frac{255 - O_t}{\left[(Voise)^2 + \frac{1}{12} \right]^{\frac{1}{2}}}$$

where the offset (O_t) , effectively the mean count in space, and the noise, effectively the standard deviation of the count in space, are calculated from



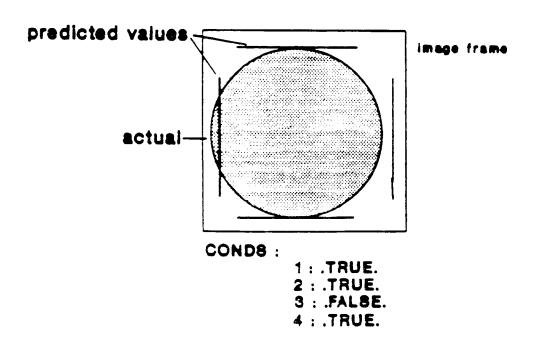


Figure 22. Image conditions (CONDS).

$$O_r = \frac{1}{N_r} \sum_{K=0}^{255} KF_r(K)$$

Noise =
$$\left[\frac{1}{N_r} \sum_{K=0}^{255} K^2 F_r(K) - O_r^2\right]^{\frac{1}{2}}$$

where K is the pixel value, F_i is the histogram of the space corners, $F_i(K)$ is the number of pixels with value K and N_i is the total number of pixels under consideration. This calculation of the signal to noise ratio includes all the quantisation noise and can be considered to be the most pessimustic S/N ratio. A more optimistic value would be given by

$$S_r N (opt) = \frac{255 - O_r}{Noise}$$

In reality the true value will be somewhere between these two values.

SNNOM: contains the signal-to-noise ratios in the space corners for IR1 or IR2, VIS1 or VIS4, VIS2 or VIS3, WV1 or WV2.

SNNLIN: contains the number of lines used to calculate SNNOM

SNREP (SNRWP): gives the S/N ratios for the eastern (western) space corners of the image for IR1 or IR2, VIS1 or VIS4, VIS2 or VIS3, WV1 or WV2.

SWMNEP (SWMNWP): gives the mean count of the eastern (western) space corners of the image for IR1 or IR2, VIS1 or VIS4, VIS2 or VIS3, WV1 or WV2.

SNMXEP (SNMXWP): gives the maximum space count of the eastern (western) space corners of the image for IR1 or IR2, VIS1 or VIS4, VIS2 or VIS3, WV1 or WV2.

7.5 Administrative messages

Offset	Name	Туре	Length	Content
560	ADMMSG	ASCII	900	Niministrative message

Table 6. Interpretation data block: administrative message format.

Up to 800 ASCII characters can be present in the imministrative message. These are used to inform users about changes or disruptions to the minimum NI II OSAT operations. The messages are marked with a senal number which consists of the scar and month of issue and a sequence number indicating the number of the message within the invinting.

8.0 METEOSAT DISSEMINATION SCHEDULE

METEOSAT Dissemination Schedules list all formats disseminated from METEOSAT for reception by PDUS and SDUS (Secondary Data User Station used for receiving WEFAX formats). An example of such a schedule, which can be provided to users on request to EUMETSAT, is shown in Figure 23.

The spacecraft has two dedicated dissemination channels which can be used for both analogue and digital transmissions. Commencing with the start of operations with METEOSAT-4, on 19 June 1989, all digital dissemination formats (A. B and X-formats) will be transmitted on Channel A2 (1694.5 MHz). Existing users should note that this is a change to the previous policy under which digital data formats were transmitted on Channel A1 (1691.0 MHz). Channel A2 also includes all GOES WEFAX transmissions from CMS Lannion, some METEOSAT formats and transmissions of processed data e.g. the Cloud Top Height product and a small number of weather charts. Channel A1 is dedicated to the dissemination of METEOSAT WEFAX images. In addition, administrative messages and test patterns are transmitted at regular intervals throughout the day in WEFAX format on both channels.

In order to standardise the start time of individual formats, four minute transmission slots are used. Formats start at the nominal times of H + 02, H + 06, H + 10 etc., with the actual transmission scheduled to commence a few seconds after these nominal times. High Resolution formats may occupy between one (BI-format) and six (AV-format) of these transmission slots. The formats do not precisely occupy the total time allocated for their transmission and on Channel A2 there is no uplink in the gap between the end of one HR format transmission and the start of the next time slot. The situation is different on Channel A1, where gaps between WEFAX formats are used for the retransmission of messages from Data Collection Platforms (DCP).

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SOME FORMATS ARE SUBJECT TO SUPPRESSION DUE TO SEASONAL UNDERILLUMINATION

Figure 23. Example of the METEOSAT Dissemination Schedule

8.1 Key to the METEOSAT Dissemination Schedule

The following notes provide information on interpreting the dissemination schedule.

SCHEDULE NUMBER

The schedule number is in the form NYYMMONN

```
YY = year of introduction.
MM = month of introduction.
C = satellite indicator (= M).
NN = number of schedule introduced
in this month.
```

Revised schedules are distributed to users in advance of their date of validity and are also announced by administrative message.

COLUMN HEADINGS

- CH Al is the METEOSA'l dissemination channel at 1691.0 MHz.
- CH A2 is the METEOSAT dissemination channel at 1694.5 MHz.
- HH is the hour during which the transmission commences.

ROW HEADINGS

- MM gives the minute of the hour at which the transmission will commence.

TABLE ENTRIES

These entries indicate the format type and slot number of the image disseminated. From this information the user can determine the area of the image, the spectral channels and the end time of the image to be disseminated. The slot number is simply related to to the end time of the image as follows:

```
slot 24 = image ending near to 1200 UTC.
slot 39 = image ending near to 1930 UTC.
```

i.e. the slot number is twice the end time of the image in hours (UTC)

The formats transmitted from CMS I annion are all prefixed with an 'L. For digital data formats the first letter indicates the area of the format and the following letters the channels to be disseminated, in accordance with the descriptions given in Section 4 of this document. Slots containing

the entry 'TEST' indicate the transmission of a test pattern, those with the entry 'ADMIN' indicate the transmission of an administrative message, both in WEFAX format.

Examples

- AV 18 = full resolution visible image of the full disk as seen by METEOSAT for slot 18.
- BIVW 32 = infrared plus visible plus water vapour image of the European and North Atlantic area for slot 32.
- LXI 19 = infrared image of area 'X' as seen by the GOES-E spacecraft for slot 19.

8.2 Availability of High Resolution formats

HR formats are distributed at regular intervals throughout the day in accordance with the dissemination schedule agreed with EUMETSAT. The dissemination schedule introduced on 19 June 1989 provides for an 'A' and 'B' format for each image slot to be disseminated approximately every half an hour. An 'X' format is disseminated approximately every three hours. The spectral channels disseminated are dependent on the time of day e.g. the visible channel data is only disseminated when the Earth's disk is sufficiently illuminated.

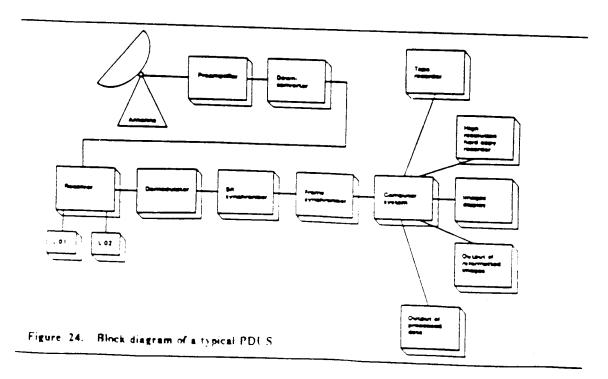
Users should note that if an image(s) is not available for dissemination, then the most recently available image in the sequence slot-1, slot-2, slot-3, will be disseminated in its place.

9.0 TECHNICAL SPECIFICATION OF A PRIMARY DATA USER STATION (PDUS)

The basic components of a PDUS are the ground station and the image handling system. The essential elements of the ground station are an antenna, low noise amplifier, down converter, receiver, demodulator and bit and frame synchronisers. The image handling system varies according to the requirements of each individual user but will require at least a computer and magnetic storage device, typically this could be either digital tape decks or Winchester disks with a minimum storage capability of 80 megabytes. Some possible options which may be required are:

- Colour image display;
- High resolution hardcopy recorder;
- Link to a larger computer system for image processing;
- Video recorder.

An example of a PDUS configuration is shown in Figure 24.



9.1 PDUS System Recommendations

The following PDUS system recommendations specifications are intended as guidelines for prospective users, however they are not complete since a PDUS will depend on user requirements and location of the station. If further advice is required the user may consult EUMETSAT.

The antenna must be suitable for operation in the 1.7 GHz band; a front-fed parabolic reflector with a diameter of 4.5 m is recommended. The receiver consists of a low noise front-end, one or more down converters and the demodulator. The selection of the input frequency of the receiver has to be made by the user but typically a frequency of 135 MHz is suitable. Since the possibility exists that HR dissemination may take place on either of the two METEOSAT dissemination channels it may be advantageous to forsec two switchable input frequencies. The recommended bandwidth of the receiver is 1 MHz and the demodulator output will be a PCM signal with a bit rate of 166.66 kbits/s.

Bit synchronisers with bit rates from 100 bits s to 5 Mbits/s and frame synchronisers with word lengths of 8 to 24 bits and frame lengths of 4 to 256 words are commonly available; the selection of the word length will depend on the computer to be used.

The recommended G/T ratio for a PDUS is 10.5dB.K⁻¹; this figure can be realised with different system concepts. For instance, the use of a high quality antenna with a diameter of 4.5 m (antenna gain 33 dB) permits the use of a low cost transistor amplifier with a noise figure of approximately 100 K.

Modulation specifications are as follows:

Modulation PCM (SP-L)/PM Index 1.2 rad +/- 5%

Bit rate 166.66 kbits/s

RF/F 660 Hz

Min. signal to noise

ratio (C/No) for

receiver -67.0 dBHz

For further information please contact

The Director

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